

UNCLASSIFIED

AD NUMBER

**AD809330**

NEW LIMITATION CHANGE

TO

**Approved for public release, distribution  
unlimited**

FROM

**Distribution authorized to U.S. Gov't.  
agencies and their contractors; Critical  
Technology; DEC 1966. Other requests shall  
be referred to Air Force Flight Dynamics  
Lab., Attn: FDTR. Wright-Patterson AFB, OH  
45433.**

AUTHORITY

**AFFDL ltr, 12 Mar 1973**

THIS PAGE IS UNCLASSIFIED

AFFDL-TR-68-109

30  
33  
36  
38  
39  
40

**TRANSIENT ANALYSIS OF HEAT CONDUCTION  
THROUGH A SLAB BY INFINITE SERIES**

THOMAS N. BERNSTEIN  
ROBERT M. ENGLE, JR.

TECHNICAL REPORT AFFDL-TR-68-109

DECEMBER 1966

This document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Theoretical Mechanics Branch, Structures Division, Air Force Flight Dynamics Laboratory (FDTR), Wright-Patterson Air Force Base, Ohio 45433.

AIR FORCE FLIGHT DYNAMICS LABORATORY  
RESEARCH AND TECHNOLOGY DIVISION  
AIR FORCE SYSTEMS COMMAND  
WRIGHT-PATTERSON AIR FORCE BASE, OHIO

## NOTICES

When Government drawings, specifications, or other data are used for any purpose other than in connection with a definitely related Government procurement operation, the United States Government thereby incurs no responsibility nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data, is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use, or sell any patented invention that may in any way be related thereto.

Copies of this report should not be returned to the Research and Technology Division unless return is required by security considerations, contractual obligations, or notice on a specific document.

**Best  
Available  
Copy**

AFFDL-TR-66-109

**TRANSIENT ANALYSIS OF HEAT CONDUCTION  
THROUGH A SLAB BY INFINITE SERIES**

**THOMAS N. BERNSTEIN  
ROBERT M. ENGLE, JR.**

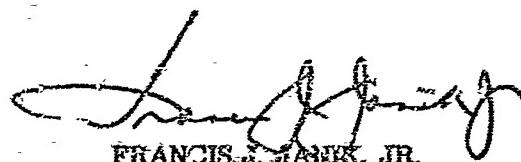
This document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Theoretical Mechanics Branch, Structures Division, Air Force Flight Dynamics Laboratory (FDTR), Wright-Patterson Air Force Base, Ohio 45433.

FOREWORD

This report was prepared by Thomas N. Bernstein and Robert M. Engle, Jr., of the Theoretical Mechanics Branch, Structures Division, Air Force Flight Dynamics Laboratory. The work was conducted in house under Project No. 1467, "Structural Analysis Methods," Task No. 146702, "Thermoelastic Structural Analysis Methods," and was administered by the Air Force Flight Dynamics Laboratory, Research and Technology Division, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio. Mr. Robert M. Bader is the Project Engineer administering Project No. 1467.

This report covers research conducted from July 1964 to July 1966. The manuscript was released by the authors in September 1966 for publication as a technical report.

This technical report has been reviewed and is approved.



FRANCIS J. TANEKE, JR.  
Chief, Theoretical Mechanics Branch  
Structures Division

ABSTRACT

The exact solution to the problem of conduction of heat through a slab is developed. The solution, formulated in terms of an infinite series, allows arbitrary initial conditions and time-dependent boundary conditions. The solution is programmed in FORTRAN IV for the IBM 7094 II computer. Several check problems were solved and the results were compared with those obtained from a finite difference heat transfer program.

PREVIOUS PAGE WAS BLANK, THEREFORE WAS NOT FED.

AFFDL-TR-66-109

## CONTENTS

| SECTION   | PAGE |
|---|------|
| I. INTRODUCTION   | 1    |
| II. MATHEMATICAL FORMULATION  | 1    |
| A. Boundary Conditions  | 1    |
| B. Steady State Solution  | 2    |
| C. Transient Solution   | 3    |
| D. Modification of Solution for Time Dependent Boundary Conditions                  | 6    |
| E. Solution of Initial Condition Problem  | 7    |
| F. Unsteady State Solution  | 7    |
| G. Transient Solution   | 7    |
| H. Complete Solution for the General Problem  | 9    |
| III. APPLICATIONS   | 10   |
| IV. COMPUTER PROGRAM FOR SERIES TRANSIENT ANALYSIS OF SLAB<br>HEAT TRANSFER (STASH) | 11   |
| A. Description  | 11   |
| B. Input  | 13   |
| C. Sample Problem   | 19   |
| D. Restrictions   | 21   |
| E. Output   | 22   |
| V. CONCLUSIONS  | 26   |
| APPENDIX I COMPUTER PROGRAM SOURCE LISTING  | 27   |
| APPENDIX II EIGENVALUE SUBROUTINES  | 41   |
| APPENDIX III RESULTS OF CHECK PROBLEMS  | 45   |

## ILLUSTRATIONS

| FIGURE  | PAGE |
|---|------|
| 1. Geometrical Representation                           | 2    |
| 2. Simplified Flow Chart of Transfer of Information     | 12   |
| 3. Input Data Format                                    | 17   |
| 4. Sample Problem Data Deck                             | 20   |
| 5. Sample Output (Case 4) Showing Display of Input Data | 23   |
| 6. Sample Output (Case 4) Showing Eigenvalue Solution   | 24   |
| 7. Sample Output (Case 4) Showing Intermediate Print    | 25   |
| 8. Temperature Profiles (Case 1)                        | 50   |
| 9. Temperature Profiles (Case 2)                        | 51   |
| 10. Temperature Profiles (Case 3)                       | 52   |
| 11. Temperature Profiles (Case 4)                       | 53   |
| 12. Temperature Profiles (Case 5)                       | 54   |

## TABLES

| TABLE                                | PAGE |
|--------------------------------------|------|
| I Systems of Units Stored Internally | 15   |
| II Comparison of Data for Case 1     | 46   |
| III Comparison of Data for Case 2    | 47   |
| IV Comparison of Data for Case 3     | 48   |
| V Comparison of Data for Case 4      | 49   |
| VI Comparison of Data for Case 5     | 49   |

## SYMBOLS

| <u>MATH SYMBOL</u>         | <u>FORTRAN SYMBOL</u> | <u>PHYSICAL DEFINITION</u>            |
|----------------------------|-----------------------|---------------------------------------|
| $A, \bar{A}, A_0, A_L$     |                       | Coefficients of Steady State Solution |
| $B, \bar{B}, B_0, B_L$     |                       | Coefficients of Steady State Solution |
| $C_0, C_1, C_2, C_n$       |                       | Constants of Integration              |
| $C_p$                      | CP                    | Specific heat                         |
| $D$                        | DETK                  | Determinant of $K_{ij}$ 's            |
| $E_0, E_L$                 | EO, EL                | boundary condition constants          |
| $f(x), F_x$                | FOFX(X)               | Initial conditions                    |
| $K$                        | K                     | Thermal conductivity                  |
| $K_{ij}$                   | K11, K21, ...         | Boundary condition indicators         |
| $L$                        | L                     | Length                                |
| $n$                        | NTERMS                | Summation index                       |
| $N$                        |                       | Particular value of n                 |
| $t$                        | T                     | Time                                  |
| $T$                        | TEMP                  | Temperature                           |
| $T_s, T_c$                 |                       | Steady state solutions                |
| $T_T, T_L, T_{Ic}, T_{IL}$ |                       | Transient solutions                   |
| $T_{IC}$                   |                       | Solution of initial condition problem |
| $T$                        |                       | Complete problem solution             |
| $S$                        |                       | Cross-sectional area of slab          |
| $x$                        | X                     | Distance                              |
| $X(x)$                     |                       | Assumed solution                      |

## SYMBOLS (Cont'd)

| <u>MATH SYMBOL</u>            | <u>FORTRAN SYMBOL</u> | <u>PHYSICAL DEFINITION</u>        |
|-------------------------------|-----------------------|-----------------------------------|
| $\gamma_n$                    |                       | Repetitive term in solution       |
| $Z, Z_n$                      | ZN                    | Eigenvalues                       |
| $\infty$                      |                       | Infinity                          |
| $\alpha = K/\rho C_p$         | ALPHA                 | Thermal diffusivity               |
| $\beta, \beta_n$              | BETAN                 | Eigenvalues                       |
| $\lambda$                     | LAMBDA                | Dummy time variable               |
| $\pi$                         | PI                    | 3.1415926                         |
| $\rho$                        | RHO                   | Density                           |
| $\phi_0(t)$                   | PHIO(T)               | Boundary condition time functions |
| $\phi_L(t)$                   | PHIL(T)               |                                   |
| $\Phi(t)$                     |                       | Assumed solution                  |
| <u>Subscripts</u>             |                       |                                   |
| IC                            |                       | Initial condition                 |
| L, O                          |                       | Boundary                          |
| P                             |                       | Pressure                          |
| S                             |                       | Steady state                      |
| T                             |                       | Transient                         |
| 0, 1, 2, 3, i, n, N           |                       | Counters                          |
| <u>Superscripts</u>           |                       |                                   |
| Primes denote differentiation |                       |                                   |

## SECTION I INTRODUCTION

The conduction of heat through a slab is governed by the following partial differential equation:

$$\frac{\partial}{\partial x} \left[ K \frac{\partial T}{\partial x} \right] = \rho C_p \frac{\partial T}{\partial t} \quad (1)$$

For constant thermal diffusivity, this equation simplifies to

$$\frac{\partial^2 T}{\partial x^2} = \frac{1}{\alpha} \frac{\partial T}{\partial t} \quad (2)$$

The exact solution to this equation can be formulated in terms of an infinite series. This report develops the exact solution for arbitrary initial conditions and time dependent boundary conditions. The solution has been programmed in FORTRAN for an IBM 7094 computer and the source program listing is contained in Appendix L.

## SECTION II MATHEMATICAL FORMULATION

### A. BOUNDARY CONDITIONS

The general solution of Equation (2) must satisfy arbitrary initial and time dependent boundary conditions which can be expressed in the following form:

$$T(x, t) = f(x) \quad t = 0 \quad (3)$$

$$K_{11} \frac{\partial T}{\partial x} + K_{12} T = F_0 \phi_0 (\lambda) \quad x = 0 \quad (4)$$

$$K_{21} \frac{\partial T}{\partial x} + K_{22} T = F_L \phi_L (\lambda) \quad x = L \quad (5)$$

where the  $K_{ij}$ 's are constants. Selecting different values of these coefficients dictates the mode of heat transfer present at the boundary. By various combinations of constants, the imposition of surface temperature, convection, heat flux or insulation is possible. A more detailed discussion on the interpretation of boundary conditions is contained in Section III, "Applications."

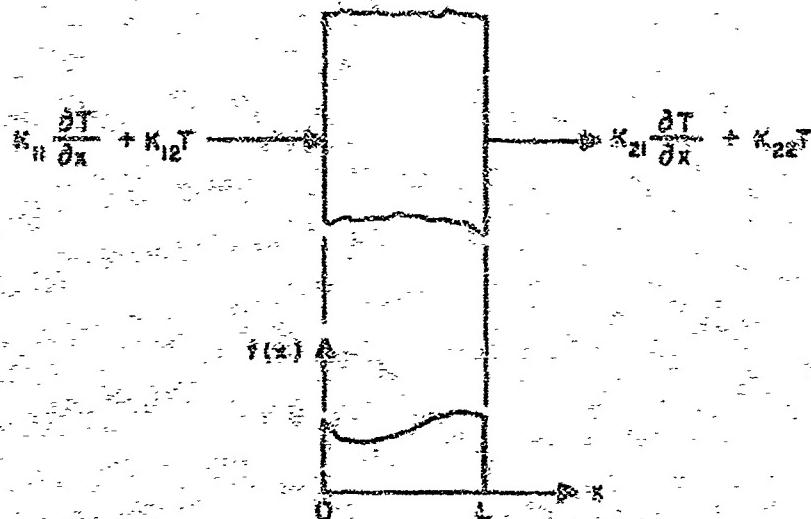


Figure 1. Geometrical Representation

The application of Duhamel's superposition theorem to account for the time dependent boundary conditions necessitated breaking the solution into a transient portion satisfying initial conditions, and a steady state plus transient with zero initial conditions.

Since the restriction has been placed on the boundary conditions that the time dependency can be expressed as a product of a time function and one of the standard boundary conditions, it is therefore possible to solve the equations neglecting the time variation and then modify the solution to account for it.

The problem is first simplified by breaking the solution into two parts: a steady state portion,  $T_s(x, \infty)$ , satisfying the arbitrary boundary condition, and a transient portion,  $T_t(x, t)$ , satisfying the initial temperature distribution and homogeneous boundary conditions.

### B. STEADY STATE SOLUTION

For steady state conditions we note that  $\frac{\partial T}{\partial t} = 0$  and Equation (3) simplifies to

$$\frac{\partial^2 T}{\partial x^2} = 0 \quad (6)$$

The solution of this equation is found directly by integration with the result:

$$T_s = Ax + B \quad (7)$$

We now impose the arbitrary boundary conditions

$$K_{11} \frac{\partial T}{\partial x} + K_{12} T = f_1(x) \quad x = 0 \quad (8)$$

$$K_{21} \frac{\partial T}{\partial x} + K_{22} T = F_L \quad x = L \quad (9)$$

Substituting Equation (7) into (8) and (9) yields

$$K_{11} A + K_{12} B = F_0 \quad (10)$$

$$K_{21} A + K_{22}(AL + B) = F_L \quad (11)$$

The constants of integration, A and B, can now be evaluated from Equations (10) and (11). In order to keep these expressions in general form, the solution is accomplished by Cramer's rule with the result

$$A = \frac{K_{12} F_L - K_{22} F_0}{K_{12} K_{22} L - (K_{11} K_{22} - K_{12} K_{21})} \quad (12)$$

$$B = \frac{(K_{21} + K_{22} L) F_0 - K_{11} F_L}{K_{12} K_{22} L - (K_{11} K_{22} - K_{12} K_{21})} \quad (13)$$

### C. TRANSIENT SOLUTION

A product form of solution is assumed for Equation (2), and designated by  $X(x)\Phi(t)$ . Substitution into Equation (2) yields

$$X''\Phi + \frac{1}{a} X\Phi' = 0 \quad (14)$$

Rearranging

$$\frac{X''}{X} = -\frac{\Phi'}{a\Phi} \quad (15)$$

which requires that each of these functions be equivalent to some, as yet arbitrary constant. Then setting this constant equal to  $-\beta^2$  results in two ordinary differential equations of the form

$$\Phi' + a\beta^2 \Phi = 0 \quad (16)$$

$$X'' + \beta^2 X = 0 \quad (17)$$

Equation (16) has the exponential form of solution

$$\Phi = C_0 e^{-a\beta^2 t} \quad (18)$$

whereas Equation (17) is satisfied by

$$X(x) = C_1 \cos \beta x + C_2 \sin \beta x \quad (19)$$

The solution to Equation (2) is then

$$T(x,t) \approx [C_1 \cos \beta x + C_2 \sin \beta x] [C_0 e^{-\alpha \beta^2 t}] \quad (20)$$

This transient solution must satisfy the initial temperature distribution and homogeneous boundary conditions as follows:

$$T(x,t) = f(x) \quad t = 0 \quad (3)$$

$$K_{12} \frac{\partial T}{\partial x} + K_{12} T = 0 \quad x = 0 \quad (21)$$

$$K_{22} \frac{\partial T}{\partial x} + K_{22} T = 0 \quad x = L \quad (22)$$

Note first that the constant  $C_0$  can be eliminated since its effect can be included in  $C_1$  and  $C_2$ . To evaluate the remaining constants substitute Equation (20) into Equations (3), (21) and (22).

Substitution of Equation (20) into (21) yields

$$K_{12} C_1 + K_{11} \beta = 0 \quad (23)$$

from which we obtain

$$C_1 = -\frac{K_{11} \beta}{K_{12}} C_2 \quad (24)$$

At this point it becomes necessary to impose the artificial restriction that  $K_{12} \neq 0$ , in order that calculations performed on the computer remain bounded.

Substitution of Equation (20) into (22) yields

$$[K_{22} \cos \beta L - K_{21} \sin \beta L] C_1 + [K_{21} \beta \cos \beta L + K_{22} \sin \beta L] C_2 = 0 \quad (25)$$

To obtain a nontrivial solution for  $C_1$  and  $C_2$ , the determinant of their coefficients must be set equal to zero. This yields the following transcendental equation,

$$\tan z = \frac{D L z}{K_{21} K_{11} z^2 + K_{22} K_{12} L^2} \quad (26)$$

where

$$z = \beta L \quad (27)$$

and

$$D = K_{11} K_{22} - K_{12} K_{21} \quad (28)$$

Equation (26) has infinitely many solutions (eigenvalues), and we shall denote these by  $\beta_n$ , where  $n = 0, 1, 2 \dots$ . The remaining constant  $C_2$  is evaluated by substituting our solution into Equation (3) in order that the initial temperature distribution be satisfied. It is obvious at this point that, in general, arbitrary functions for the initial temperature distribution can not be satisfied using only one value of  $Z_n$  and  $C_2$ . We are thus required to expand our initial condition and our solution in an infinite series. Then the coefficient  $C_2$  becomes  $C_n$  and its evaluation proceeds as follows. The total solution at this point can be expressed

$$T(x, t) = Ax + B + \sum_{n=0}^{\infty} C_n \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right] e^{-\alpha \beta_n^2 t} \quad (29)$$

Imposing the problem initial condition yields

$$T(x, 0) = f(x) = Ax + B + \sum_{n=0}^{\infty} C_n \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right] \quad (30)$$

Rearranging

$$f(x) - (Ax + B) = \sum_{n=0}^{\infty} C_n \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right] \quad (31)$$

Since the sines and cosines form a complete set of orthogonal functions, the  $C_n$ 's can be evaluated by multiplying both sides of Equation (31) by

$$\left[ \sin \beta_N x - \frac{K_{11} \beta_N}{K_{12}} \cos \beta_N x \right] \text{ and integrating from zero to } L.$$

Thus,

$$\int_0^L \left[ f(x) - (Ax + B) \right] \left[ \sin \beta_N x - \frac{K_{11} \beta_N}{K_{12}} \cos \beta_N x \right] dx = \int_0^L \sum_{n=0}^{\infty} C_n \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right] \left[ \sin \beta_N x - \frac{K_{11} \beta_N}{K_{12}} \cos \beta_N x \right] dx \quad (32)$$

By orthogonality this integration produces nontrivial results only in the case of  $n = N$ . Therefore

$$C_n = \frac{\int_0^L \left[ f(x) - (Ax + B) \right] \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right] dx}{\int_0^L \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right]^2 dx} \quad (33)$$

The denominator of  $C_n$  can be evaluated directly with the result:

$$\int_0^L \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right]^2 dx = \frac{1}{2 \beta_n} \left\{ z_n \left[ \left( \frac{K_{11} \beta_n}{K_{12}} \right)^2 + 1 \right] \right. \\ \left. + \sin z_n \cos z_n \left[ \left( \frac{K_{11} \beta_n}{K_{12}} \right)^2 - 1 \right] + 2 \left( \frac{K_{11} \beta_n}{K_{12}} \right) \sin^2 z_n \right\} \quad (34)$$

Collecting the formulations required for problem evaluation leads to the expression of the solution for time independent boundary conditions in the form

$$T(x, t) = Ax + B \\ + \sum_{n=0}^{\infty} \frac{\int_0^L [f(x) - (Ax + B)] [Y_n(x)] dx}{\int_0^L [Y_n(x)]^2 dx} [Y_n(x)] e^{-\alpha \beta_n^2 t} \quad (35)$$

where

$$A = \left[ \frac{K_{12} F_L - K_{22} F_0}{K_{11} K_{22} L - 0} \right] \quad (36)$$

$$B = \left[ \frac{(K_{21} + K_{22} L) F_0 - K_{11} F_L}{K_{11} K_{22} L - 0} \right] \quad (37)$$

$$Y_n(x) = \left[ \sin \beta_n x - \frac{K_{11} \beta_n}{K_{12}} \cos \beta_n x \right] \quad (38)$$

$$\beta_n = z_n / \lambda \quad (39)$$

and

$$\tan z_n = Z_n = \frac{D L Z_n}{K_{21} K_{11} Z_n^2 + K_{22} K_{12} L^2} \quad (40)$$

#### D. MODIFICATION OF SOLUTION FOR TIME DEPENDENT BOUNDARY CONDITIONS

Duhamel's superposition integral is now applied to the solution, Equation (35), to account for the time varying boundary conditions. F. B. Hildebrand gives the solution in the form\*

$$T(x, t) = T_0(x, \omega) \phi(x) + \left\{ \phi(0) + \int_0^t e^{\alpha \beta_n^2 \lambda} \phi''(\lambda) d\lambda \right\} T_0(x) \quad (41)$$

This formulation is based on certain limitations, however, which must be eliminated. The first is the assumption of zero initial conditions. This restriction is eliminated by considering a

\*Hildebrand, F. B., Introduction to Numerical Analysis, McGraw-Hill Book Company, Inc., New York, 1956.

separate problem, possessing the given initial conditions and homogeneous boundary conditions. The solution of this problem is added to Equation (41). The second simplification utilized to obtain Equation (41) was to hold one boundary at zero and consider the remaining boundary to vary with time. For our problem, both boundaries can vary with time so we make use of the superposition principle once again by varying first one boundary condition and then the other, with the remaining boundary held at zero. The two results are then added. Note that  $f(x) = 0$  for both these solutions.

#### E. SOLUTION OF INITIAL CONDITION PROBLEM

For the given initial condition, Equation (3), and the homogeneous boundary conditions, Equations (21) and (22), a "zero" steady state solution is obtained from Equation (7), i.e.,  $A = 0$ ,  $B = 0$ . Employing the given initial condition in Equation (35) then yields the desired result,

$$T_{IC}(x,t) = \sum_{n=1}^{\infty} \frac{\int_0^L f(x) [Y_n(x)] dx}{\int_0^L [Y_n(x)]^2 dx} [Y_n(x)] e^{-\alpha \beta_n^2 t} \quad (42)$$

#### F. UNSTEADY STATE SOLUTION

The steady state solution,  $T_s(x,\infty)$ , employed in Equation (35) is modified to  $T_s(x,\infty)\phi(t)$  in Equation (41). This result can be viewed as the steady state solution to a problem with our boundary conditions, if those conditions were "frozen" at the instant  $t$ , and remained constant as  $t \rightarrow \infty$ . Since our boundary conditions vary continuously with time, it is not possible to reach a steady state condition. This explains the title employed for this section of the report.  $T_s(x,\infty)\phi(t)$  can be obtained immediately from Equation (7), using boundary condition Equations (4) and (5) in place of (8) and (9). The result is

$$T_s \phi(t) = T_0 = Ax + B \quad (43)$$

where

$$A = \frac{K_{21} F_L \phi_L(t) - K_{22} F_0 \phi_0(t)}{K_{12} K_{22} L - D} \quad (44)$$

$$B = \frac{(K_{21} + K_{22} L) F_0 \phi_0(t) - K_{11} F_L \phi_L(t)}{K_{12} K_{22} L - D} \quad (45)$$

#### G. TRANSIENT SOLUTION (TIME VARIABLE BOUNDARY CONDITIONS)

The transient solution for boundary condition Equations (4) and (5) must be evaluated in two parts as indicated in Section II D. First, consider the conditions

$$K_{11} \frac{\partial T}{\partial x} + K_{12} T = F_0 \phi_0(\lambda) \quad (4)$$

and

$$K_{21} \frac{\partial T}{\partial x} + K_{22} T = 0 \quad (22)$$

with

$$f(x) = 0 \quad (46)$$

Equations (44) and (45) for  $A$  and  $B$  are modified by letting  $F_L = 0$ , with the result

$$A_0 = \frac{-K_{22}F_0 \phi_0(t)}{K_{12}K_{22}L-D} \quad (47)$$

$$B_0 = \frac{(K_{21} + K_{22}L) F_0 \phi_0(t)}{K_{12}K_{22}L-D} \quad (48)$$

The transient solution is obtained by substituting these results into the transient solution in Equation (35) with the result

$$T_{T_0} = \sum_{n=0}^{\infty} \frac{\int_0^L (-B_0 - A_0 x) Y_n(x) dx}{\int_0^L [Y_n(x)]^2 dx} \left[ Y_n(x) \right] e^{-\alpha \beta_n^2 t} \quad (49)$$

Similarly for the conditions

$$K_{11} \frac{\partial T}{\partial x} + K_{12} T = 0 \quad (50)$$

$$K_{21} \frac{\partial T}{\partial x} + K_{22} T = F_L \phi_L(t) \quad (51)$$

$$f(x) = 0 \quad (46)$$

and letting  $F_0 = 0$  in Equations (41) and (45), with the result

$$A_L = \frac{K_{12} F_L \phi_L(t)}{K_{12}K_{22}L-D} \quad (50)$$

$$B_L = \frac{-K_{11} F_L \phi_L(t)}{K_{12}K_{22}L-D} \quad (51)$$

we obtain from Equation (35) the transient solution

$$T_{T_L} = \sum_{n=0}^{\infty} \frac{\int_0^L (A_L x - B_L) Y_n(x) dx}{\int_0^L [Y_n(x)]^2 dx} \left[ Y_n(x) \right] e^{-\alpha \beta_n^2 t} \quad (52)$$

## H. COMPLETE SOLUTION FOR THE GENERAL PROBLEM

The general form of the complete solutions was expressed by Equation (41). Collecting the solutions obtained in Equations (42), (43), (49) and (52), and substituting into Equation (41), yields the final result,

$$\begin{aligned} \overline{\Psi}(x,t) = & \Psi_0 + \left\{ \phi_0(0) + \int_0^t e^{-a\beta_n^2 \lambda} \phi'_0(\lambda) d\lambda \right\} \Psi_{T_0}(x,t) \\ & + \left\{ \phi_L(0) + \int_0^t e^{-a\beta_n^2 \lambda} \phi'_L(\lambda) d\lambda \right\} \Psi_{T_L}(x,t) + \Psi_{IC}(x,t) \end{aligned} \quad (53)$$

### SECTION III APPLICATIONS

The generalized boundary conditions utilized in the mathematical formulation can be specialized to handle a number of physical problems. For example, take Equation (5)

$$K_{21} \frac{\partial T}{\partial x} + K_{22} T = F_L \phi_L(\lambda) \quad (5)$$

Appropriate choices of the indicators lead to the following:

- a. Prescribed constant surface temperature.

Let  $K_{21} = 0$ ,  $K_{22} = 1$ ,  $\phi_L(\lambda) = 1$ ,  $F_L$  = applied temperature

- b. Prescribed constant heat flux.

Let  $K_{21} = KS$ ,  $K_{22} = 0$ ,  $\phi_L(\lambda) = 1$ ,  $F_L$  = applied flux

- c. Insulated boundary.

Let  $K_{21} = KS$ ,  $K_{22} = 0$ ,  $\phi_L(\lambda) = 1$ ,  $F_L = 0$

- d. Linear heat transfer at the surface (convection).

Let  $K_{21} = KS$ ,  $K_{22} = -BS$ ,  $\phi_L(\lambda) = 1$ ,  $F_L = -hST_c$

This results in a boundary condition equation of the form

$$KS \frac{\partial T}{\partial x} = hS(T - T_c)$$

where  $h$  is the usual convective heat transfer coefficient per unit area.

- e. Sign Convention.

The sign convention is such that a positive sign indicates flux into the body.

The boundary conditions described in "a" and "b" above may be arbitrarily varied with time by applying the appropriate time function,  $\phi_L(\lambda)$ .

## SECTION IV

COMPUTER PROGRAM FOR SERIES TRANSIENT  
ANALYSIS OF SLAB HEAT TRANSFER (STASH)

## A. DESCRIPTION

The program described below was written to solve for the temperature distribution in a one-dimensional rod with arbitrary initial conditions and time-varying boundary conditions. STASH is coded in FORTRAN IV for the IBM 7044-7094 II Direct Coupled-System. Fifteen subprograms make up the program, each of which has a specific task to perform. These subprograms are listed below.

- |         |   |
|---------|---|
| MAIN    | - reads in data, sets up calculations, and prints the results.                  |
| SOLVE 1 | - solves the eigenvalue equation for positive values of DETK (see Appendix II). |
| SOLVE 2 | - solves the eigenvalue equation for negative values of DETK.                   |
| SOLVE 3 | - solves the eigenvalue equation for a zero value of DETK.                      |
| SOLVE 4 | - solves the eigenvalue equation for DETK infinite.                             |
| FINT    | - Simpson's rule integration routine.   |
| FUNCX   | - sets up the integrand for the $x$ integral                                    |
| FUNCT   | - sets up the integrand for the $\lambda$ integral                              |
| TABIN   | - reads in tabular data, if present   |
| INTERP  | - performs linear interpolation on tabular data                                 |
| PHIO    | - defines the time varying boundary condition at $x = 0$                        |
| PHIL    | - defines the time varying boundary condition at $x = L$                        |
| PHIPRO  | - defines the derivative of the time-varying boundary condition at $x = 0$      |
| PHIPRL  | - defines the derivative of the time-varying boundary condition at $x = L$      |
| FOFX    | - defines the initial conditions in the rod.                                    |

Figure 2 is a simplified flow chart depicting the transfer of information between the subprograms discussed above.

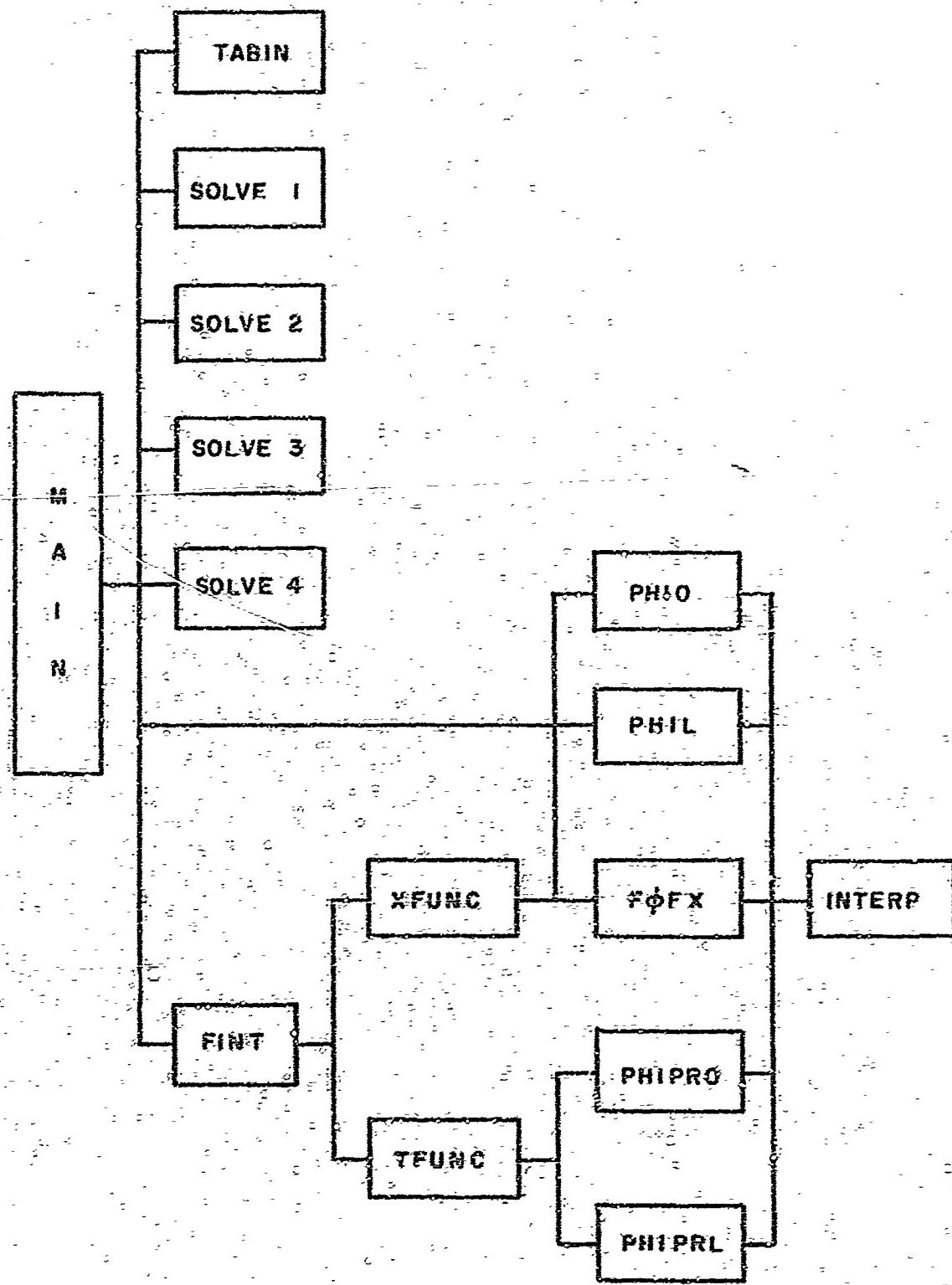


Figure 2. Simplified Flow Chart of Transfer of Information

B. INPUT

There are basically two types of input data to STASH, the physical parameters and the problem parameters. The physical parameters are the characteristics of the rod and the conditions to which it is subjected. The problem parameters are the accuracy parameters, the calculation controls and the print controls.

The rod is characterized by four basic quantities: length, mass density, thermal conductivity and specific heat. The boundary conditions are identified by the indicators  $K_{11}$ ,  $K_{12}$ ,  $K_{21}$ ,  $K_{22}$ . The magnitude of the boundary conditions is characterized by the indicators  $F_0$  and  $F_L$ . The magnitude of the initial conditions is similarly characterized by the indicator  $F_x$ .

Since the solution may be required at any station on the rod for any given point in time, it is convenient to specify a length increment and a time increment as input parameters. A final time is also specified to terminate calculations. The accuracy of the solution is basically governed by three factors: the accuracy of the eigenvalues, the number of terms in the series portion of the solution and the number of intervals taken in the numerical integration routine. In the interest of maximum flexibility, each of these quantities was made an input parameter.

Information may be input to the program in two basic forms. The first type is the data card. There will always be seven cards in the data deck. If the tabular data option is used there may be many more. The second type of input consists of FORTRAN IV statements which may be inserted into the subprograms defining the initial and boundary conditions in the rod. An example problem using both types of input is given in Section V C. Detailed instructions on inputting the data cards are given below, in the following format:

- (1) Card number and contents
- (2) Program name for contents
- (3) Format of card input referenced to format statement number
- (4) Description of each variable on the card

1. Data Cards

Card 1 Intermediate print options

JPRINT(1), JPRINT(2), JPRINT(3)

5000 FORMAT(3H1)

JPRINT(1) - prints series portion of solution term by term  
if a one is entered. If no print is desired enter a zero.

JPRINT(2) - prints unsteady state portion of the solution if a  
one is entered. If no print is desired enter a zero.

JPRINT(3) - prints solution for eigenvalues if a one is entered.  
If no print is desired enter a zero.

Card 2 Title Card

IUNIT, TITLE

1 FORMAT (12, 13A6)

IUNIT - an indicator which prints out the system of units to be used in the problem. See Table I for a list of systems presently contained in the program.

TITLE - any alphanumeric information through column 80.

Card 3 Physical Parameters (all must be in consistent units)

L, K, RHO, CP, DELTAX, DELTAT, TIMEF

2 FORMAT (7E10.0)

L length of the rod

K thermal conductivity

RHO mass density

CP specific heat

DELTAX increment of length (100 increments maximum)

DELTAT increment of time

TIMEF final time (initial time is zero)

Card 4 Boundary Condition Indicators

K11, K12, K21, K22

3 FORMAT (4E10.0)

K11 indicator for  $\frac{\partial T}{\partial x}$  at  $x = 0$

K12 indicator for T at  $x = 0$

K21 indicator for  $\frac{\partial T}{\partial x}$  at  $x = L$

K22 indicator for T at  $x = L$

Card 5 Function Multiplying Factors

FO, FL, FX

4 FORMAT (3E10.0)

FO coefficient on function  $\phi_0(t)$

FL coefficient on function  $\phi_L(t)$

FX coefficient on function f(x)

TABLE I  
SYSTEMS OF UNITS STORED INTERNALLY

| IUNIT | LENGTH | MASS  | TIME | WEIGHT | TEMPERATURE |
|-------|--------|-------|------|--------|-------------|
| 1     | INCH   | SLUG  | SEC  | POUND  | FAHRENHEIT  |
| 2     | INCH   | SLUG  | MIN  | POUND  | FAHRENHEIT  |
| 3     | INCH   | SLUG  | HR   | POUND  | FAHRENHEIT  |
| 4     | FOOT   | SLUG  | SEC  | POUND  | FAHRENHEIT  |
| 5     | FOOT   | SLUG  | MIN  | POUND  | FAHRENHEIT  |
| 6     | FOOT   | SLUG  | HR   | POUND  | FAHRENHEIT  |
| 7     | INCH   | POUND | SEC  | POUND  | FAHRENHEIT  |
| 8     | INCH   | POUND | MIN  | POUND  | FAHRENHEIT  |
| 9     | INCH   | POUND | HR   | POUND  | FAHRENHEIT  |
| 10    | FOOT   | POUND | SEC  | POUND  | FAHRENHEIT  |
| 11    | FOOT   | POUND | MIN  | POUND  | FAHRENHEIT  |
| 12    | FOOT   | POUND | HR   | POUND  | FAHRENHEIT  |

Card 6 Calculation Parameters

NTERMS, NSTEPX, NSTEPT, NTAB(1), NTAB(2), NTAB(3), NTAB(4),

NTAB(5);

5 FORMAT (SI5, 5I1)

NTERMS - number of terms in the series portion of the solution (100 maximum)

NSTEPX - number of intervals for the x-integration

NSTEPT - number of intervals for the  $\lambda$ -integration

NTAB(1) - flag for table 1

NTAB(2) - flag for table 2

NTAB(I) = 0 Do not use Table

NTAB(3) - flag for table 3

NTAB(4) - flag for table 4

NTAB(I) = 1 Use Table

NTAB(5) - flag for table 5

If NSTEPX or NSTEPT is zero the program sets the value of the respective integral to zero.

Card 7 Eigenvalue Solution Parameters

LIMIT, ITERMAX

6 FORMAT (E10.0, I5)

LIMIT - difference between two successive iterations necessary to define convergence to a root.

ITERMAX - maximum number of iterations to be made in searching for each eigenvalue.

If no tabular data is to be used, this is the last card in the data deck. If tabular data is to be an input, however, the following format will be used.

Card 8 Table Number and Comments

NTABLE, COMMENTS

FORMAT (I5, 20x, 52H )

NTABLE table number

COMMENTS any alphanumeric information in columns 26 through 80.

Card 2 Tabular Data ((2 to 50 data cards per table)

INDVAR, DEPVAR, COMMENTS

FORMAT (5X, 2E10.0, 55H )

INDVAR independent variable

DEPVAR dependent variable

COMMENTS any alphanumeric information in columns 26 through 80

Card 3 End of Table

N

FORMAT (15)

N negative of table number

There are five tables provided in the program, which are assigned as follows:

Table 1 PHIQ

Table 2 PHIL

Table 3 PHIPRO

Table 4 PHIPRL

Table 5 FOFX

Figure 3. shows a symbolic data deck. A sample problem is generated in detail in Section IV C.

2. Subprogram Input Cards

If the tabular data option is not used STAS<sup>TM</sup> evaluates the required functions internally using FORTRAN statements as loaded in the subprograms at compilation time. The affected subprograms are:

FUNCTION PHIQ

FUNCTION PHIL

FUNCTION PHIPRO

FUNCTION PHIPRL

FUNCTION FOFX

As the initial or boundary conditions change, cards containing the functional statement of the variation must be inserted. Since all the above functions have an associated multiplying factor

| STATMENT                            |                          | FORTRAN STATEMENT                         |         |        |      |
|-------------------------------------|--------------------------|---|---------|--------|------|
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | CARD 1 PRINT CONTROLS                     |         |        |      |
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | J 1=1 PRINT SERIES SOLUTION ITEM BY ITEM  |         |        |      |
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | J 1=0 DO NOT PRINT                        |         |        |      |
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | J 2=1 PRINT STEADY STATE SOLUTION         |         |        |      |
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | J 2=0 DO NOT PRINT                        |         |        |      |
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | J 3=1 PRINT EIGENVALUE SOLUTION           |         |        |      |
| <input checked="" type="checkbox"/> | <input type="checkbox"/> | J 3=0 DO NOT PRINT                        |         |        |      |
|                                     |                          | CARD 2 TITLE CARD                         |         |        |      |
| 15                                  |                          | ANY ALPHANUMERIC DATA                     |         |        |      |
| I 0                                 |                          | ENTRIES IDENTIFICATION FLAG (SEE TABLE I) |         |        |      |
|                                     |                          | CARD 3 PHYSICAL PARAMETERS                |         |        |      |
|                                     |                          | C P                                       | DELTAX  | DELTAT | TIME |
|                                     |                          | X 11                                      | X 12    | X 21   | X 22 |
|                                     |                          | CARD 4 BOUNDARY CONDITION INDICATORS      |         |        |      |
|                                     |                          | FL  | FX      |        |      |
|                                     |                          | CARD 5 FUNCTION MULTIPLYING FACTORS       |         |        |      |
|                                     |                          | LIMIT                                     | ITRMAX  |        |      |
|                                     |                          | CARD 6 CALCULATION PARAMETERS             |         |        |      |
| INTERMIST                           |                          | EQUATION                                  |         |        |      |
| K STEP                              |                          | L12345                                    |         |        |      |
|                                     |                          | CARD 7 EIGENVALUE SOLUTION PARAMETERS     |         |        |      |
| N TAB                               |                          | TABULAR DATA                              |         |        |      |
|                                     |                          | IND VAR                                   | DEP VAR |        |      |
|                                     |                          | X   | Y       |        |      |
|                                     |                          | Z   | W       |        |      |
|                                     |                          | INR VAR                                   | DEP VAR |        |      |
| END T                               |                          |   |         |        |      |

Figure 3. Input Data Format

it is convenient to use normalized functional statements in the subprogram. Thus, if we entered the following card in the FUNCTION FOFX

$$\text{FOFX} = 1.0$$

we would obtain the initial condition

$$f(x) = FX$$

The program (Appendix I) contains subprogram statements corresponding to the following initial and boundary conditions.

$$\phi_0(t) = \text{const}$$

$$\phi_L(t) = \text{const}$$

$$\phi'_0(t) = 0$$

$$\phi'_L(t) = 0$$

$$f(x) = \text{const}$$

### C. SAMPLE PROBLEM

Consider a rod, ten inches long, having the following properties:

$$K = 10 \text{ Btu/br-ft-}^{\circ}\text{F}$$

$$\rho = 500 \text{ lb}_m/\text{ft}^3$$

$$c_p = 0.1 \text{ Btu/lb}_m$$

We wish to obtain a time history of the temperature - distribution through the rod which is subject to the following boundary conditions

$$T(0, t) = t$$

$$T(L, t) = 0$$

The initial conditions imposed on the rod are:

$$T(x, 0) = 0$$

For purposes of illustration the intermediate print option will be called out on card one.

The first task in setting up the problem is to decide upon a system of units to employ. Since the length of the rod is given in inches, we shall choose the inch as the unit of length. For a transient study a small time unit is desirable, hence, the second becomes the unit of time. The remainder of the system of units is determined by the temperature and mass units. Going now to Table I, we obtain the correct value of IUNIT. This indicator along with a suitable title becomes card two in Figure 4.

FORTRAN CODING TODAY

| FORTRAN STATEMENT   |   |   |   |    |    |    |    |    |    |
|---------------------|---|---|---|----|----|----|----|----|----|
| STATEMENT<br>NUMBER | 5 | 6 | 7 | 10 | 11 | 14 | 20 | 25 | 30 |
| 1.1.1               |   |   |   |    |    |    |    |    |    |
| 1.                  |   |   |   |    |    |    |    |    |    |
| 2.                  |   |   |   |    |    |    |    |    |    |
| 3.                  |   |   |   |    |    |    |    |    |    |
| 4.                  |   |   |   |    |    |    |    |    |    |
| 5.                  |   |   |   |    |    |    |    |    |    |
| 6.                  |   |   |   |    |    |    |    |    |    |
| 7.                  |   |   |   |    |    |    |    |    |    |
| 8.                  |   |   |   |    |    |    |    |    |    |
| 9.                  |   |   |   |    |    |    |    |    |    |
| 10.                 |   |   |   |    |    |    |    |    |    |
| 11.                 |   |   |   |    |    |    |    |    |    |
| 12.                 |   |   |   |    |    |    |    |    |    |
| 13.                 |   |   |   |    |    |    |    |    |    |
| 14.                 |   |   |   |    |    |    |    |    |    |
| 15.                 |   |   |   |    |    |    |    |    |    |
| 16.                 |   |   |   |    |    |    |    |    |    |
| 17.                 |   |   |   |    |    |    |    |    |    |
| 18.                 |   |   |   |    |    |    |    |    |    |
| 19.                 |   |   |   |    |    |    |    |    |    |
| 20.                 |   |   |   |    |    |    |    |    |    |
| 21.                 |   |   |   |    |    |    |    |    |    |
| 22.                 |   |   |   |    |    |    |    |    |    |
| 23.                 |   |   |   |    |    |    |    |    |    |
| 24.                 |   |   |   |    |    |    |    |    |    |
| 25.                 |   |   |   |    |    |    |    |    |    |
| 26.                 |   |   |   |    |    |    |    |    |    |
| 27.                 |   |   |   |    |    |    |    |    |    |
| 28.                 |   |   |   |    |    |    |    |    |    |
| 29.                 |   |   |   |    |    |    |    |    |    |
| 30.                 |   |   |   |    |    |    |    |    |    |
| 31.                 |   |   |   |    |    |    |    |    |    |
| 32.                 |   |   |   |    |    |    |    |    |    |
| 33.                 |   |   |   |    |    |    |    |    |    |
| 34.                 |   |   |   |    |    |    |    |    |    |
| 35.                 |   |   |   |    |    |    |    |    |    |
| 36.                 |   |   |   |    |    |    |    |    |    |
| 37.                 |   |   |   |    |    |    |    |    |    |
| 38.                 |   |   |   |    |    |    |    |    |    |
| 39.                 |   |   |   |    |    |    |    |    |    |
| 40.                 |   |   |   |    |    |    |    |    |    |
| 41.                 |   |   |   |    |    |    |    |    |    |
| 42.                 |   |   |   |    |    |    |    |    |    |
| 43.                 |   |   |   |    |    |    |    |    |    |
| 44.                 |   |   |   |    |    |    |    |    |    |
| 45.                 |   |   |   |    |    |    |    |    |    |
| 46.                 |   |   |   |    |    |    |    |    |    |
| 47.                 |   |   |   |    |    |    |    |    |    |
| 48.                 |   |   |   |    |    |    |    |    |    |
| 49.                 |   |   |   |    |    |    |    |    |    |
| 50.                 |   |   |   |    |    |    |    |    |    |
| 51.                 |   |   |   |    |    |    |    |    |    |
| 52.                 |   |   |   |    |    |    |    |    |    |
| 53.                 |   |   |   |    |    |    |    |    |    |
| 54.                 |   |   |   |    |    |    |    |    |    |
| 55.                 |   |   |   |    |    |    |    |    |    |
| 56.                 |   |   |   |    |    |    |    |    |    |
| 57.                 |   |   |   |    |    |    |    |    |    |
| 58.                 |   |   |   |    |    |    |    |    |    |
| 59.                 |   |   |   |    |    |    |    |    |    |
| 60.                 |   |   |   |    |    |    |    |    |    |
| 61.                 |   |   |   |    |    |    |    |    |    |
| 62.                 |   |   |   |    |    |    |    |    |    |
| 63.                 |   |   |   |    |    |    |    |    |    |
| 64.                 |   |   |   |    |    |    |    |    |    |
| 65.                 |   |   |   |    |    |    |    |    |    |
| 66.                 |   |   |   |    |    |    |    |    |    |
| 67.                 |   |   |   |    |    |    |    |    |    |
| 68.                 |   |   |   |    |    |    |    |    |    |
| 69.                 |   |   |   |    |    |    |    |    |    |
| 70.                 |   |   |   |    |    |    |    |    |    |
| 71.                 |   |   |   |    |    |    |    |    |    |
| 72.                 |   |   |   |    |    |    |    |    |    |
| 73.                 |   |   |   |    |    |    |    |    |    |
| 74.                 |   |   |   |    |    |    |    |    |    |
| 75.                 |   |   |   |    |    |    |    |    |    |
| 76.                 |   |   |   |    |    |    |    |    |    |
| 77.                 |   |   |   |    |    |    |    |    |    |
| 78.                 |   |   |   |    |    |    |    |    |    |
| 79.                 |   |   |   |    |    |    |    |    |    |
| 80.                 |   |   |   |    |    |    |    |    |    |
| 81.                 |   |   |   |    |    |    |    |    |    |
| 82.                 |   |   |   |    |    |    |    |    |    |
| 83.                 |   |   |   |    |    |    |    |    |    |
| 84.                 |   |   |   |    |    |    |    |    |    |
| 85.                 |   |   |   |    |    |    |    |    |    |
| 86.                 |   |   |   |    |    |    |    |    |    |
| 87.                 |   |   |   |    |    |    |    |    |    |
| 88.                 |   |   |   |    |    |    |    |    |    |
| 89.                 |   |   |   |    |    |    |    |    |    |
| 90.                 |   |   |   |    |    |    |    |    |    |
| 91.                 |   |   |   |    |    |    |    |    |    |
| 92.                 |   |   |   |    |    |    |    |    |    |
| 93.                 |   |   |   |    |    |    |    |    |    |
| 94.                 |   |   |   |    |    |    |    |    |    |
| 95.                 |   |   |   |    |    |    |    |    |    |
| 96.                 |   |   |   |    |    |    |    |    |    |
| 97.                 |   |   |   |    |    |    |    |    |    |
| 98.                 |   |   |   |    |    |    |    |    |    |
| 99.                 |   |   |   |    |    |    |    |    |    |
| 100.                |   |   |   |    |    |    |    |    |    |
| 101.                |   |   |   |    |    |    |    |    |    |
| 102.                |   |   |   |    |    |    |    |    |    |
| 103.                |   |   |   |    |    |    |    |    |    |
| 104.                |   |   |   |    |    |    |    |    |    |
| 105.                |   |   |   |    |    |    |    |    |    |
| 106.                |   |   |   |    |    |    |    |    |    |
| 107.                |   |   |   |    |    |    |    |    |    |
| 108.                |   |   |   |    |    |    |    |    |    |
| 109.                |   |   |   |    |    |    |    |    |    |
| 110.                |   |   |   |    |    |    |    |    |    |
| 111.                |   |   |   |    |    |    |    |    |    |
| 112.                |   |   |   |    |    |    |    |    |    |
| 113.                |   |   |   |    |    |    |    |    |    |
| 114.                |   |   |   |    |    |    |    |    |    |
| 115.                |   |   |   |    |    |    |    |    |    |
| 116.                |   |   |   |    |    |    |    |    |    |
| 117.                |   |   |   |    |    |    |    |    |    |
| 118.                |   |   |   |    |    |    |    |    |    |
| 119.                |   |   |   |    |    |    |    |    |    |
| 120.                |   |   |   |    |    |    |    |    |    |
| 121.                |   |   |   |    |    |    |    |    |    |
| 122.                |   |   |   |    |    |    |    |    |    |
| 123.                |   |   |   |    |    |    |    |    |    |
| 124.                |   |   |   |    |    |    |    |    |    |
| 125.                |   |   |   |    |    |    |    |    |    |
| 126.                |   |   |   |    |    |    |    |    |    |
| 127.                |   |   |   |    |    |    |    |    |    |
| 128.                |   |   |   |    |    |    |    |    |    |
| 129.                |   |   |   |    |    |    |    |    |    |
| 130.                |   |   |   |    |    |    |    |    |    |
| 131.                |   |   |   |    |    |    |    |    |    |
| 132.                |   |   |   |    |    |    |    |    |    |
| 133.                |   |   |   |    |    |    |    |    |    |
| 134.                |   |   |   |    |    |    |    |    |    |
| 135.                |   |   |   |    |    |    |    |    |    |
| 136.                |   |   |   |    |    |    |    |    |    |
| 137.                |   |   |   |    |    |    |    |    |    |
| 138.                |   |   |   |    |    |    |    |    |    |
| 139.                |   |   |   |    |    |    |    |    |    |
| 140.                |   |   |   |    |    |    |    |    |    |
| 141.                |   |   |   |    |    |    |    |    |    |
| 142.                |   |   |   |    |    |    |    |    |    |
| 143.                |   |   |   |    |    |    |    |    |    |
| 144.                |   |   |   |    |    |    |    |    |    |
| 145.                |   |   |   |    |    |    |    |    |    |
| 146.                |   |   |   |    |    |    |    |    |    |
| 147.                |   |   |   |    |    |    |    |    |    |
| 148.                |   |   |   |    |    |    |    |    |    |
| 149.                |   |   |   |    |    |    |    |    |    |
| 150.                |   |   |   |    |    |    |    |    |    |
| 151.                |   |   |   |    |    |    |    |    |    |
| 152.                |   |   |   |    |    |    |    |    |    |
| 153.                |   |   |   |    |    |    |    |    |    |
| 154.                |   |   |   |    |    |    |    |    |    |
| 155.                |   |   |   |    |    |    |    |    |    |
| 156.                |   |   |   |    |    |    |    |    |    |
| 157.                |   |   |   |    |    |    |    |    |    |
| 158.                |   |   |   |    |    |    |    |    |    |
| 159.                |   |   |   |    |    |    |    |    |    |
| 160.                |   |   |   |    |    |    |    |    |    |
| 161.                |   |   |   |    |    |    |    |    |    |
| 162.                |   |   |   |    |    |    |    |    |    |
| 163.                |   |   |   |    |    |    |    |    |    |
| 164.                |   |   |   |    |    |    |    |    |    |
| 165.                |   |   |   |    |    |    |    |    |    |
| 166.                |   |   |   |    |    |    |    |    |    |
| 167.                |   |   |   |    |    |    |    |    |    |
| 168.                |   |   |   |    |    |    |    |    |    |
| 169.                |   |   |   |    |    |    |    |    |    |
| 170.                |   |   |   |    |    |    |    |    |    |
| 171.                |   |   |   |    |    |    |    |    |    |
| 172.                |   |   |   |    |    |    |    |    |    |
| 173.                |   |   |   |    |    |    |    |    |    |
| 174.                |   |   |   |    |    |    |    |    |    |
| 175.                |   |   |   |    |    |    |    |    |    |
| 176.                |   |   |   |    |    |    |    |    |    |
| 177.                |   |   |   |    |    |    |    |    |    |
| 178.                |   |   |   |    |    |    |    |    |    |
| 179.                |   |   |   |    |    |    |    |    |    |
| 180.                |   |   |   |    |    |    |    |    |    |
| 181.                |   |   |   |    |    |    |    |    |    |
| 182.                |   |   |   |    |    |    |    |    |    |
| 183.                |   |   |   |    |    |    |    |    |    |
| 184.                |   |   |   |    |    |    |    |    |    |
| 185.                |   |   |   |    |    |    |    |    |    |
| 186.                |   |   |   |    |    |    |    |    |    |
| 187.                |   |   |   |    |    |    |    |    |    |
| 188.                |   |   |   |    |    |    |    |    |    |
| 189.                |   |   |   |    |    |    |    |    |    |
| 190.                |   |   |   |    |    |    |    |    |    |
| 191.                |   |   |   |    |    |    |    |    |    |
| 192.                |   |   |   |    |    |    |    |    |    |
| 193.                |   |   |   |    |    |    |    |    |    |
| 194.                |   |   |   |    |    |    |    |    |    |
| 195.                |   |   |   |    |    |    |    |    |    |
| 196.                |   |   |   |    |    |    |    |    |    |
| 197.                |   |   |   |    |    |    |    |    |    |
| 198.                |   |   |   |    |    |    |    |    |    |
| 199.                |   |   |   |    |    |    |    |    |    |
| 200.                |   |   |   |    |    |    |    |    |    |
| 201.                |   |   |   |    |    |    |    |    |    |
| 202.                |   |   |   |    |    |    |    |    |    |
| 203.                |   |   |   |    |    |    |    |    |    |
| 204.                |   |   |   |    |    |    |    |    |    |
| 205.                |   |   |   |    |    |    |    |    |    |
| 206.                |   |   |   |    |    |    |    |    |    |
| 207.                |   |   |   |    |    |    |    |    |    |
| 208.                |   |   |   |    |    |    |    |    |    |
| 209.                |   |   |   |    |    |    |    |    |    |
| 210.                |   |   |   |    |    |    |    |    |    |
| 211.                |   |   |   |    |    |    |    |    |    |
| 212.                |   |   |   |    |    |    |    |    |    |
| 213.                |   |   |   |    |    |    |    |    |    |
| 214.                |   |   |   |    |    |    |    |    |    |
| 215.                |   |   |   |    |    |    |    |    |    |
| 216.                |   |   |   |    |    |    |    |    |    |
| 217.                |   |   |   |    |    |    |    |    |    |
| 218.                |   |   |   |    |    |    |    |    |    |
| 219.                |   |   |   |    |    |    |    |    |    |
| 220.                |   |   |   |    |    |    |    |    |    |
| 221.                |   |   |   |    |    |    |    |    |    |
| 222.                |   |   |   |    |    |    |    |    |    |
| 223.                |   |   |   |    |    |    |    |    |    |
| 224.                |   |   |   |    |    |    |    |    |    |
| 225.                |   |   |   |    |    |    |    |    |    |
| 226.                |   |   |   |    |    |    |    |    |    |
| 227.                |   |   |   |    |    |    |    |    |    |
| 228.                |   |   |   |    |    |    |    |    |    |
| 229.                |   |   |   |    |    |    |    |    |    |
| 230.                |   |   |   |    |    |    |    |    |    |
| 231.                |   |   |   |    |    |    |    |    |    |
| 232.                |   |   |   |    |    |    |    |    |    |
| 233.                |   |   |   |    |    |    |    |    |    |
| 234.                |   |   |   |    |    |    |    |    |    |
| 235.                |   |   |   |    |    |    |    |    |    |
| 236.                |   |   |   |    |    |    |    |    |    |
| 237.                |   |   |   |    |    |    |    |    |    |
| 238.                |   |   |   |    |    |    |    |    |    |
| 239.                |   |   |   |    |    |    |    |    |    |
| 240.                |   |   |   |    |    |    |    |    |    |
| 241.                |   |   |   |    |    |    |    |    |    |
| 242.                |   |   |   |    |    |    |    |    |    |
| 243.                |   |   |   |    |    |    |    |    |    |
| 244.                |   |   |   |    |    |    |    |    |    |
| 245.                |   |   |   |    |    |    |    |    |    |
| 246.                |   |   |   |    |    |    |    |    |    |
| 247.                |   |   |   |    |    |    |    |    |    |
| 248.                |   |   |   |    |    |    |    |    |    |
| 249.                |   |   |   |    |    |    |    |    |    |

Figure 4. Sample Problem Data Deck

Card three contains the physical parameters of the system in the system of units called for by the indicator, IUNIT. These parameters are:

$$L = 10 \text{ in. } K = 0.002314 \text{ Btu/in.-sec-}^{\circ}\text{F} \quad \rho = 0.2895 \text{ lb}_m/\text{in.}^3$$

$$CP = 0.10 \text{ Btu/lb}_m \quad DELTAX = 0.5 \text{ in.} \quad DELTAT = 100 \text{ sec} \quad TIME = 1000 \text{ sec}$$

Since we have only temperature boundary conditions we set  $K11 = K21 = 0$  and  $K12 = K22 = 1$ . This information is entered on card four.

The functions defining our initial conditions are

$$\phi_0(t) = t, \phi_L(t) = 0, f(x) = 0, \phi'_0(\lambda) = 1, \phi'_L(\lambda) = 0$$

We can take advantage of the functions already stored in the program by using  $F_L$  and  $F_X$  to zero out the appropriate functions. We then can use  $F_0 = 1$  to bring in the other boundary condition. This is shown on card five.

The calculation parameters are entered on card six. A series solution containing twenty-five terms is completely adequate for this problem. Since the initial condition function is zero, we set NSTEPX equal to zero. For a Simpson's rule integration scheme fifty steps should suffice for NSTEPT. Our choice of multiplying factors on card five enabled us to do much of the function calculation internally. However, to eliminate recompiling any portion of the program we used tables to define the functions  $\phi_0(t)$  and  $\phi'_0(\lambda)$ . Thus we set NTAB(1) and NTAB(3) equal to one and the rest equal to zero.

For a solution with temperature boundaries only, the eigenvalues become simply  $n\pi$ . Thus, the eigenvalue parameters have little meaning. However, for a more complex case they would have a significant effect on the solution so these parameters should be made as stringent as required. Typical values are entered on card seven.

Our choice not to recompile any subprograms leads to the use of tables one and three. The first card in each table is a table designation number. The following cards contain data points. The last card of each table contains the negative of the table designation number and is a flag signalling the end of the table.

This, then is the data deck for the sample problem. The assembled deck is shown in Figure 4.

#### D. RESTRICTIONS

Certain restrictions must be adhered to in order for the solution to be successful. Violation of these restrictions will usually produce an error message from the computer program.

- A consistent set of units must be employed. An indicator is provided on the title card which will label the system of units on the output. If this indicator is omitted, the following error message is printed:

SYSTEM OF UNITS NOT SPECIFIED. IUNIT NOT ENTERED OR ZERO.

This message merely informs the user of this omission, execution of the problem is not terminated.

b.  $K_{11}$  and  $K_{12}$  cannot be zero simultaneously. Similarly,  $K_{21}$  and  $K_{22}$  cannot be zero simultaneously. These situations lead to an undefined boundary. The error message below results from this case.

BOTH INDICATORS AT ONE BOUNDARY ARE ZERO

c.  $K_{12}$  cannot be zero. This is a somewhat artificial restriction imposed by the formulation of the problem. In a physical sense it prevents the possibility of the unsolvable Neuman problem. If  $K_{12}$  is zero the following error message is printed.

FORMULATION DOES NOT PERMIT K12 TO BE ZERO.

d. The number of integration intervals must be even. This restriction arises from the computer formulation of Simpson's rule. If an uneven number is entered, the following error message results:

NUMBER OF INTEGRATION INTERVALS IS NOT EVEN

e. The computer program generates an error message if the time increment or the length increment is zero or negative. This message reads:

TIME OR LENGTH INCREMENT IS ZERO OR NEGATIVE

f. The initial time for each program is zero..  
g. The program uses even increments of time and length. The maximum number of length increments is one hundred.

E. OUTPUT

The output generated by STASH consists of two segments; the input data display and temperature profiles which are always generated; and the intermediate print which is controlled by the first card in the data deck. After reading the data, STASH prints it out along with suitable titles and headings as shown in Figure 5. If the eigenvalue solution is requested a table of the eigenvalues, and iterations, is printed as shown in Figure 6. Intermediate print options giving the values of the series and the unsteady state produce output as shown in Figure 7 for each station along the slab at each time step. Figure 7 also shows the form of the temperature profiles as generated at each time step.

## CASE 4 INSULATED HEATPIPE

PHYSICAL CONSTANTS TO DEFINE THE PROBLEM

## SYSTEM OF UNITS

| LENGTH<br>INCH         | MASS<br>POUND    | TIME<br>SEC       | TEMPERATURE<br>FAHRENHEIT |
|------------------------|------------------|-------------------|---------------------------|
| 0.23144000E-02         | 0.28450000E-00   | 0.09999999E-01    | 0.35000000E-04            |
| INITIAL<br>CONDUTIVITY | SPECIFIC<br>HEAT | INCREMENT<br>TIME | FINAL<br>TIME             |
| 0.09999999E-12         | 0.09999999E-00   | 0.09999999E-03    | 0.09999999E-03            |

## ACUITY CONDITION INDICATORS FROM THE DIFFERENTIAL EQUATION

|     |                |                |    |
|-----|----------------|----------------|----|
| R11 | K12            | N21            | 0. |
| 0.  | 0.09999999E-01 | 0.09999999E-01 | 0. |

## MULTIPLYING FACTORS

FOR  
MOMENTUM AND INITIAL CONDITION FUNCTIONS

|                |      |      |                |
|----------------|------|------|----------------|
| R101           | F101 | R102 | R103           |
| 0.09999999E-04 | 0.   | 0.   | 0.09999999E-03 |

## CALCULATION PARAMETERS

|                                     |  |  |
|-------------------------------------|--|--|
| NUMBER OF TERMS<br>IN THE SUMMATION | NUMBER OF INTERVALS<br>FOR THE X INTEGRATION | NUMBER OF INTERVALS<br>FOR THE T INTEGRATION |
| 100                                 | 100  | 100  |

## EIGENVALUE SOLUTION PARAMETERS

|                   |                |
|-------------------|----------------|
| ACCURACY<br>LIMIT | ALPHA          |
| 0.99999999E-08    | 0.77930015E-01 |

## DETAILED OUTPUT

0.

0.

0.

-0.09999999E-01

Figure 6. Sample Output (Case 4) Showing Display of Input

SOLUTION FOR EIGENVALUES

ROOT NO

| ROOT NO | Z              |
|---------|----------------|
| 1       | 0.15707962E 01 |
| 2       | 0.47123838E 01 |
| 3       | 0.78539814E 01 |
| 4       | 0.10995574E 02 |
| 5       | 0.14137166E 02 |
| 6       | 0.17278759E 02 |
| 7       | 0.20420352E 02 |
| 8       | 0.23561944E 02 |
| 9       | 0.26703537E 02 |
| 10      | 0.29845123E 02 |

Figure 6. Sample Output (Case 4) Showing Eigenvalue Solution

SERIES PARTITION OF SOLUTION

## **UNSTEADY STATE POSITION OF CONFINEMENT**

Figure 7. Sample output (Case 4) showing best fit.

## SECTION V CONCLUSIONS

Several classes of problems were run to check out the program. The results were compared with a finite-difference heat transfer program (LTA) which was developed by Lockheed Aircraft Corporation. These results were examined for accuracy and speed of convergence.

The intermediate print feature of the program was used to determine the convergence. Examination showed that convergence was quite rapid, usually less than ten terms, for constant boundary conditions, away from time zero. More terms were needed in the vicinity of zero time to produce convergence. For time-varying boundary conditions, the series does not converge very quickly. However, a study of the solution convergence showed that a twenty term series using one hundred integration steps obtained results within one percent of the LTA solution for times exceeding one hundred seconds. At smaller times a fifty term series with one hundred integration steps was required.

Appendix III contains the results of five check problems compared with the results from the LTA finite difference program.

No comparison is made at zero time since the program obtains these values from the initial conditions rather than from a series calculation. The curves are plotted from data generated by the program. Case I is the sample problem detailed in Section IV C. All other cases used the same physical parameters.

AFFDL-TR-66-109

**APPENDIX I**  
**COMPUTER PROGRAM SOURCE LISTING**

```

888JCB STASH   FAP          STASH001
$IBFTC MAIN   #94/2,XR7      STASH002
C
C   A GENERAL SOLUTION TO THE ONE-DIMENSIONAL HEAT TRANSFER PROBLEM      STASH003
C   WITH TIME-DEPENDENT BOUNDARY CONDITIONS AND ARBITRARY INITIAL      STASH004
C   CONDITION      STASH005
C   VERSION 2      STASH006
C   VERSION 2 CALCULATES ZERO TIME USING BOUNDARY AND INITIAL CONDITIONS      STASH007
C   VERSION 2 INCORPORATES A MULTIPLIER ON THE INITIAL CONDITION FUNCTION      STASH008
C   AND BOUNDARY CONDITION FUNCTIONS TO PERMIT THE USE OF NORMALIZED      STASH009
C   FUNCTIONS      STASH010
C
C   EXTERNAL TFUNC,XFUNC      STASH011
C   DIMENSION TEMP(100),TITLE(13),EIGEN(100)      STASH012
C   DIMENSION JPRINT(3)      STASH013
C   DIMENSION NTAB(5)      STASH014
C   REAL K11BN,K11BN2      STASH015
C   REAL K11K21,K12K22      STASH016
C   REAL LTHIT,L,K,LAPBDA,NUXSS,NUHSS,KFPHI,KFPHIO,KFPHIL,K11,K12      STASH017
C   L,K21,K22,N,TERMH1,TERM2      STASH018
C   COMMON T,X,LAMBDA,TERMH1,TERM2,E2PHTA,BETAN,EXPHL,CBNX,SHMX,DELTA      STASH019
C   LX,DELTAT,K11,K12,K21,K22,FX,NTAB      STASH020
C   COMMON/ROCTS/ITERMS,L,ITERMX,K11K21,K12K22,LIKIT      STASH021
C   COMMON/PRINT/JPRINT      STASH022
C
C   FORMAT STATEMENTS      STASH023
C
1  FORMAT(12,13A6)      STASH024
2  FORMAT(7E10.0)      STASH025
3  FORMAT(4E10.0)      STASH026
4  FORMAT(3E10.0)      STASH027
5  FORMAT(3IS,5I1)      STASH028
6  FORMAT(1E0,0,15)      STASH029
11 FORMAT(1H1,29X,13A6)      STASH030
20 FORMAT(1H0,8X,6HLENGTH,12X,7HTHERMAL,11X,7HDENSITY,10X,8HSPECIFIC,STASH031
111X,6HLENGTH,13X,4HTIME,14X,5HFINAL/24X,12HCONDUCTIVITY,28X,4HHEAT      STASH032
2,11X,9HINCREMENT,9X,9HINCREMENT,12X,4HTIME)      STASH033
22 FORMAT(1H0,3X,E15.8,3X,E15.0,4X,E15.8,2X,E15.8,313X,E15.8))      STASH034
30 FORMAT(1HA,35X,60HBOUNDARY CONDITION INDICATORS FROM THE DIFFERENT      STASH035
11AL EQUATION//36X,3HK11,16X,3HK12,16X,3HK21,16X,3HK22)      STASH036
31 FORMAT(1HA,55X,19HMULTIPLYING FACTORS/6+X,3HFOR/45X,40HBOUNDARY AN      STASH037
10 INITIAL CONDITION FUNCTIONS//30X,4HF(0),30X,4HF(L),30X,4HF(X)//2      STASH038
24X,2(E15.8,19X),E15.8)      STASH039
33 FORMAT(1H0,29X,4(E15.8,4X))      STASH040
66 FORMAT(1HA,51X,30HEIGENVALUE SOLUTION PARAMETERS//42X,8HACCURACY,3      STASH041
*0X,9HNUMBER OF/44X,5HLIMIT,31X,10HITERATIONS//39X,E15.8,26X,15)      STASH042
120 FORMAT(28X,4HINCH,14X,4HSLUG,14X,3HSEC,16X,5HPOUND,13X,10HFARENHE      STASH043
11T)      STASH044
130 FORMAT(28X,4HINCH,14X,4HSLUG,14X,3HMIN,16X,5HPOUND,13X,10HFARENHE      STASH045
11T)      STASH046
140 FORMAT(28X,4HINCH,14X,4HSLUG,15X,2HHR,16X,5HPOUND,13X,10HFARENHE      STASH047
11T)      STASH048
150 FORMAT(28X,4HFCT,14X,4HSLUG,14X,3HSEC,16X,5HPOUND,13X,10HFARENHE      STASH049
11T)      STASH050
160 FORMAT(28X,4HFCT,14X,4HSLUG,14X,3HMIN,16X,5HPOUND,13X,10HFARENHE      STASH051
11T)      STASH052
170 FORMAT(28X,4HFOOT,14X,4HSLUG,15X,2HHR,16X,5HPOUND,13X,10HFARENHE      STASH053
11T)      STASH054
180 FORMAT(28X,4HINCH,14X,5HPOUND,13X,3HSEC,16X,5HPOUND,13X,10HFARENHE      STASH055
11T)      STASH056
190 FORMAT(28X,4HINCH,14X,5HPOUND,13X,3HMIN,16X,5HPOUND,13X,10HFARENHE      STASH057
11T)      STASH058
      STASH059
      STASH060
      STASH061
      STASH062

```

```

303 FORMAT(1H0,54Y,2ZH CALCULATION PARAMETERS//26X,1SH NUMBER OF TERMS,1STASH063
15X,19H NUMBER OF INTERVALS,13X,19H NUMBER OF INTERVALS//25X,16R IN THE STASH064
2 SUMMATION,14X,21H FOR THE X INTEGRATION,11X,21H FOR THE T INTEGRATION1STASH065
3CN//30X,15,28X,15,26X,15) STASH066
333 FORMAT(1H0,63X,5H ALPHA) STASH067
1002 FORMAT(26X,5E15.8) STASH068
1003 FORMAT(1X,E15.8) STASH069
1100 FORMAT(28X,4H INC,14X,5H POUND,14X,2H HR,16X,5H POUND,13X,10H FAHRENHESTASH070
1IT) STASH071
1110 FORMAT(28X,4H FCOT,14X,5H POUND,13X,3H SEC,16X,5H POUND,13X,10H FAHRENHESTASH072
1EIT) STASH073
1120 FORMAT(28X,4H FCOT,14X,5H POUND,13X,3H MIN,16X,5H POUND,13X,10H FAHRENHESTASH074
1EIT) STASH075
1130 FORMAT(28X,4H FCOT,14X,5H POUND,14X,2H HR,16X,5H POUND,13X,10H FAHRENHESTASH076
1IT) STASH077
1140 FORMAT(1H0,45X,40H PHYSICAL CONSTANTS TO DEFINE THE PROBLEM//59X,15STASH078
1H SYSTEM OF UNITS//27X,6H LENGTH,13X,4H MASS,14X,4H TIME,14X,6H WEIGHT,STASH079
213X,1H TEMPERATURE) STASH080
2002 FORMAT(7X,4H TIME,22X,42H TEMPERATURE DISTRIBUTION ALONG THE ROD AT STASH081
1 ,FS. 3,10H INTERVALS) STASH082
3003 FORMAT(1H0) STASH083
3333 FORMAT(1H0,58X,E15.8) STASH084
4064 FORMAT(1H0,63X,4H CTK) STASH085
4444 FORMAT(1H0,58X,E15.8) STASH086
5000 FORMAT(5E1) STASH087
6100 FORMAT(1H1,25X,2HT=,E15.8,2U4,2HX#,E15.8//45X,26H SERIES PORTION OF STASH088
* SOLUTION//14X,1HN,16X,4H TERA,19X,9H SUMMATION//) STASH089
6110 FORMAT(10X,15,10X,E15.8,10X,E15.8) STASH090
6200 FORMAT(1H0,40X,34H UNSTEADY STATE PORTION OF SOLUTION//35X,1HX,25X,STASH091
*4H TEMP//) STASH092
6210 FORMAT(28X,F15.8,20X,E15.8) STASH093
1ERROR=0 STASH094
C STASH095
C INTERMEDIATE PRINT OPTIONS STASH096
C STASH097
C READ(5,5000) JPRINT STASH098
C STASH099
C JPRINT=1, PRINT INTERMEDIATE CALCULATIONS. JPRINT=0, DO NOT PRINT. STASH100
C JPRINT(1)-SERIES PORTION OF THE SOLUTION TERM BY TERM STASH101
C JPRINT(2)-UNSTEADY STATE PORTION OF THE SOLUTION STASH102
C JPRINT(3)-SOLUTION FOR THE EIGENVALUES STASH103
C STASH104
C READ (5,1) IUNIT,(TITLE(I),I=1,13) STASH105
C READ(5,2)L,K,RHO,CP,DELTAX,DELTAT,TIMEF STASH106
C READ(5,3)K11,K12,K21,K22 STASH107
C READ(5,4)FO,FL,FX STASH108
C READ(5,5) NTERMS,NSTEPX,NSTEPT,NTAB STASH109
C READ(5,6) LIMIT,ITERMX STASH110
C STASH111
C PRINT INPUT DATA STASH112
C STASH113
C TITLE STASH114
C STASH115
C WRITE (6,2)(TITLE(I),I=1,13) STASH116
C STASH117
C SYSTEM OF UNITS STASH118
C STASH119
C WRITE(6,1140) STASH120
C IF(IUNIT.EQ.0) GO TO 5500 STASH121
C GO TO (102,103,134,105,106,107,108,109,110,111,112,113),IUNIT STASH122
102 WRITE(6,120) STASH123
STASH124

```

|      |  |          |
|------|--|----------|
|      | GO TO 9999                                       | STASH125 |
| 103  | WRITE(6,130)                                     | STASH126 |
|      | GO TO 9999                                       | STASH127 |
| 104  | WRITE(6,140)                                     | STASH128 |
|      | GO TO 9999                                       | STASH129 |
| 105  | WRITE(6,150)                                     | STASH130 |
|      | GO TO 9999                                       | STASH131 |
| 106  | WRITE(6,160)                                     | STASH132 |
|      | GO TO 9999                                       | STASH133 |
| 107  | WRITE(6,170)                                     | STASH134 |
|      | GO TO 9999                                       | STASH135 |
| 108  | WRITE(6,180)                                     | STASH136 |
|      | GO TO 9999                                       | STASH137 |
| 109  | WRITE(6,190)                                     | STASH138 |
|      | GO TO 9999                                       | STASH139 |
| 110  | WRITE(6,1100)                                    | STASH140 |
|      | GO TO 9999                                       | STASH141 |
| 111  | WRITE(6,1110)                                    | STASH142 |
|      | GO TO 9999                                       | STASH143 |
| 112  | WRITE(6,1120)                                    | STASH144 |
|      | GO TO 9999                                       | STASH145 |
| 113  | WRITE(6,1130)                                    | STASH146 |
|      | GO TO 9999                                       | STASH147 |
| 5500 | WRITE(6,5550)                                    | STASH148 |
| C    |  | STASH149 |
| C    | PHYSICAL CONSTANTS                               | STASH150 |
| C    |  | STASH151 |
| 9999 | WRITE(6,20)                                      | STASH152 |
|      | WRITE(6,22)L,K,RHO,CP,DELTA,X,DELTAT,TIMEF       | STASH153 |
| C    |  | STASH154 |
| C    | BOUNDARY CONDITIONS                              | STASH155 |
| C    |  | STASH156 |
|      | WRITE(6,30)                                      | STASH157 |
|      | WRITE(6,33)K11,K12,K21,K22                       | STASH158 |
| C    |  | STASH159 |
| C    | MULTIPLYING FACTORS                              | STASH160 |
| C    |  | STASH161 |
|      | WRITE(6,31)F0,FL,FX                              | STASH162 |
| C    |  | STASH163 |
| C    | CALCULATION PARAMETERS                           | STASH164 |
| C    |  | STASH165 |
|      | WRITE(6,203)TERMS,NSTEPX,NSTEPT                  | STASH166 |
| C    |  | STASH167 |
| C    | EIGENVALUE SOLUTION PARAMETERS                   | STASH168 |
| C    |  | STASH169 |
|      | WRITE(6,66) LIMIT,IYERMX                         | STASH170 |
| C    |  | STASH171 |
| C    | TEST FOR TABULAR DATA                            | STASH172 |
| C    |  | STASH173 |
|      | DO 50 I=1,5                                      | STASH174 |
|      | IF(INTAB(I).NE.0) CALL TABIN(I)                  | STASH175 |
| 50   | CONTINUE   | STASH176 |
| C    |  | STASH177 |
| C    | IF TABLES ARE USED, THEY ARE ASSIGNED AS FOLLOWS | STASH178 |
| C    |  | STASH179 |
| C    | NO.1 FUNCTION PHIBET                             | STASH180 |
| C    | NO.2 FUNCTION PHILIT                             | STASH181 |
| C    | NO.3 FUNCTION PHIPBD(LAMBDA)                     | STASH182 |
| C    | NO.4 FUNCTION PHIPBL(LAMBDA)                     | STASH183 |
| C    | NO.5 FUNCTION FOFX(X)                            | STASH184 |
| C    |  | STASH185 |
|      |  | STASH186 |

```

C          ERROR CHECKS ON INPUT DATA                      STASH187
C
C          NUMBER OF INTEGRATION STEPS MUST BE EVEN OR ZERO.    STASH188
C          IF ZERO THE PROGRAM SETS THE INTEGRAL EQUAL TO ZERO. STASH189
C
C          IF(NSTEPX.EQ.0) GO TO 248                         STASH190
C          IF(MOD(NSTEPX,2).EQ.0) GO TO 248                  STASH191
C          WRITE(6,5553)                                     STASH192
C          IERROR=IERROR+1                                    STASH193
C
248      IF(NSTEPY.EQ.0) GO TO 249                         STASH194
C          IF(MOD(NSTEPY,2).EQ.0) GO TO 249                  STASH195
C          WRITE(6,5553)                                     STASH196
C          TERROR=TERROR+1                                    STASH197
C
249      IF((IDELTAX.GT.0.))                                STASH198
*          .AND.                                         STASH199
*          (DELTAZ.GT.0.))                                STASH200
*          GO TO 250                                       STASH201
C          WRITE(6,5551)                                     STASH202
C          IERROR=IERROR+1                                    STASH203
C
250      IF(TIMEF.GT.0.) GO TO 251                         STASH204
C          WRITE(6,5552)                                     STASH205
C          IERROR=IERROR+1                                    STASH206
C
C          K12 CANNOT BE ZERO DUE TO A RESTRICTION IN THE FORMULATION. STASH207
C
251      IF (K12.NE.0.) GO TO 252                         STASH208
C          WRITE(6,5555)                                     STASH209
C          IERROR=IERROR+1                                    STASH210
C
C          CHECK FOR UNDEFINED BOUNDARY CONDITIONS.           STASH211
C
252      IF(((K11.NE.0.).OR.(K12.NE.0.)))                 STASH212
*          .AND.                                         STASH213
*          ((K21.NE.0.).OR.(K22.NE.0.)))                 STASH214
*          GO TO 253                                       STASH215
C          WRITE(6,441)                                     STASH216
C          IERROR=IERROR+1                                    STASH217
C
253      IF(IERROR.GT.0) STOP                           STASH218
C
C          WRITE(6,333)                                     STASH219
C          PI=3.1415926                                  STASH220
C          ALPHA=K/(RHO*CP)                            STASH221
C          WRITE(6,3333) ALPHA                          STASH222
C          DETK=K11*K22-K12*K21                      STASH223
C          WRITE(6,4004)                               STASH224
C          WRITE(6,4444) DETK                          STASH225
C
C          DETERMINE CONSTANTS FOR USE IN DO LOOPS          STASH226
C
C          CONOM=K12*K22*L-DETK                         STASH227
C          CONO1=K11*X12*L                               STASH228
C          CONO2=K11**2                                 STASH229
C          CONO3=(K12*L)**2                            STASH230
C          CONO4=FL                                    STASH231
C          CONO5=K11*K21                             STASH232
C          CONO6=K12*K22*(L**2)                         STASH233
C          CONO7=DETK*L                               STASH234
C          CONO8=F0                                    STASH235
C          CONO9=K12*FL                               STASH236
C
C
C
C          CONOM=K12*K22*L-DETK                         STASH237
C          CONO1=K11*X12*L                               STASH238
C          CONO2=K11**2                                 STASH239
C          CONO3=(K12*L)**2                            STASH240
C          CONO4=FL                                    STASH241
C          CONO5=K11*K21                             STASH242
C          CONO6=K12*K22*(L**2)                         STASH243
C          CONO7=DETK*L                               STASH244
C          CONO8=F0                                    STASH245
C          CONO9=K12*FL                               STASH246
C
C
C          CONOM=K12*K22*L-DETK                         STASH247
C          CONO1=K11*X12*L                               STASH248
C

```

```

CON10=K22*F0          STASH249
CON11=F0/K12          STASH250
CON12=K12*DFNOM       STASH251
CON088=K12**2         STASH252
K11K21=K11*K21        STASH253
K12K22=K12*K22        STASH254
C
C DETERMINING EIGENVALUES FOR SERIES SOLUTION      STASH255
C
C IF((K11K21.EQ.0.).AND.(K12K22.EQ.0.))GO TO 400   STASH256
IF(DET.LT.250,300,100  STASH257
100 CALL SOLVE1(DETK,EIGEN,NTERMS)    STASH258
GO TO 900             STASH259
200 CALL SOLVE2(DETK,EIGEN,NTERMS)    STASH260
GO TO 900             STASH261
300 CALL SOLVE3(DETK,EIGEN,NTERMS)    STASH262
GO TO 900             STASH263
400 CALL SOLVE4(DETK,EIGEN,NTERMS)    STASH264
C
C SET UP INDICES FOR DO LOOPS                      STASH265
C
900 NTEMPT=(TIMFF/DELTAT)+1.5          STASH266
ATEMPX=(L/DELTAX)+1.5          STASH267
WRITE(6,3003)           STASH268
T=-DELTAT              STASH269
DO 199 NT=1,NTEMPT      STASH270
T=T+DELTAT              STASH271
X=-DELTAX               STASH272
DO 999NX=1,NTEMPX       STASH273
IF(T.EQ.0.) GO TO 299    STASH274
X=X+DELTAX              STASH275
SERIES=0.                STASH276
IF(JPRINT(I).NE.0) WRITE(6,6100) T,X      STASH277
C
C SET UP LOOP TO GENERATE SERIES SUMMATION FOR TRANSIENT SOLUTION  STASH278
C
C0999I=1,NTERMS          STASH279
ZN=EIGEN(I)              STASH280
9  GETAN=ZN/L              STASH281
ZN2=ZN**2                 STASH282
ZDENOM=ZN2*DENOM          STASH283
SBNX=SIN(BETAN*X)        STASH284
CBNX=COS(BETAN*X)        STASH285
SZN=SIN(ZN)               STASH286
CZN=COS(ZN)               STASH287
K11BN=K11*BETAN          STASH288
K11BN2=K11BN**2          STASH289
ZN2SZN=ZN2*SZN            STASH290
EXPHUL=EXP(-(ALPHA*(BETAN)**2)*T)        STASH291
SUMCON=BETAN*(K12*SBNX-K11BN*CBNX)/(ZN*(K11BN2+CON088)+(K11BN2-  STASH292
ICON088)*ZN*CZN-2.*K11BN*K12*(ZN**2))  STASH293
KTERM1=((CON01*ZN2SZN-CON02*ZN2SZN+CON03*(ZN*CZN-SZN))/ZDENOM)*  STASH294
ICON04                  STASH295
KTERM2=((CON05*ZN2SZN+CON06*(SZN-ZN)+CON07*ZN*(1.-CZN))/ZDENOM)*  STASH296
ICON08                  STASH297
KFPHIL=KTERM1*PHIL(0.)    STASH298
KFPHI0=KTERM2*PHI0(0.)    STASH299
KFPHI=EXPHUL*(KFPHI0+KFPHIL)  STASH300
IF(NSTEPT.EQ.0) GOTO 399    STASH301
TIMINT=FIAT(0.,T,NSTEPT,TFUNC)  STASH302
GO TO 599                STASH303
399 TIMINT=0.              STASH304

```

```

DATA PI/3.1415926/
SLOPE =DET K/(K12K22*L)
IF(JPRINT(3).NE.0) WRITE(6,6300)
C0999 I=1,ITERMS
NN=I
IFI K11K21.EQ.0.) GO TO 801
GO TO 799
801 IF(ABS(SLCPE).LT.1.0) GO TO 800
799 Z0=(2.*NN-1.)*PI/2.)
GO TO 850
800 Z0=(2.*NN+1.)*PI/2.)
850 BOUND=0.0
CO 99 J=1,ITERMX
U=DET K=L*Z0/(K11K21+Z0**2+K12K22*L**2)
IF(ABS(SLOPE).LT.1.0) GO TO 860
Z1=(NN-1.)*PI+ATAN(U)
GO TO 870
860 Z1=NN*PI+ATAN(U)
870 IF(ABS(Z1-Z0).LT.LIMIT) GO TO 9
IF(BOUND.EQ.(Z1-Z0)) GO TO 90
BOUND=Z1-Z0
Z0=Z1
IF(JPRINT(3).NE.0) WRITE(6,6310) I,J,U,Z1
99 CONTINUE
WRITE(6,11) I,ITERMX
STOP
9 EIGEN(I)=Z1
GO TO 999
90 EIGEN(I)=(Z1+Z0)/2.0
999 CONTINUE
RETURN

C
C          ERROR MESSAGES
C
11 FORMAT(10X,12HROOT NUMBER ,I3,22H010 NOT CONVERGE AFTER,I5,11H ITERATIONS)
END
66
$1BFTC SOLV2 #94/2,XRT
SUBROUTINE SOLVE2(DETK,EIGEN,ITERMS)
C
C      SOLVES TAN(Z)=DETL*Z/(K11*K21+Z**2+K22*K12*L**2)
C
DIMENSION EIGEN(ITERMS)
DIMENSION JPRINT(3)
DIMENSION NTAB(5)
REAL K11K21,K12K22,L,LIMIT,NN
REAL LIIMIT,L,K,LAM8DA,NUMXSS,NUMS3,KFPH1,KFPH10,KEPHIL,K11,K12
L,K21,K22,N,KTERM1,KTERM2
COMMON T,X,LAM8DA,KTERM1,KTERM2,ALPHA,BETAK,EXPNU1,CBNX,SBNX,DELTA
IX,DELTAT,K11,K12,K21,K22,GX,NTAB
COMMON/X/ROTS/ITERMS,L,ITERMX,K11K21,K12K22,LIMIT
COMMON/PRINT/JPRINT
6300 FORMAT(1H1,46X,24HSOLUTION FOR EIGENVALUES//26X,7HROOT NO,5X,9HITERMS)
*RATION,10X,1HU,24X,1HZ//)
6310 FORMAT(28X,I3,10X,I3,6X,E15.8,10X,E15.8)
DATA PI/3.1415926/
IFI JPRINT(3).NE.0) WRITE(6,6300)
C0999 I=1,ITERMS
NN=I
Z0=(NN+1.)*PI
850 BOUND=0.0
STASH373
STASH374
STASH375
STASH376
STASH377
STASH378
STASH379
STASH380
STASH381
STASH382
STASH383
STASH384
STASH385
STASH386
STASH387
STASH388
STASH389
STASH390
STASH391
STASH392
STASH393
STASH394
STASH395
STASH396
STASH397
STASH398
STASH399
STASH400
STASH401
STASH402
STASH403
STASH404
STASH405
STASH406
STASH407
STASH408
STASH409
STASH410
STASH411
STASH412
STASH413
STASH414
STASH415
STASH416
STASH417
STASH418
STASH419
STASH420
STASH421
STASH422
STASH423
STASH424
STASH425
STASH426
STASH427
STASH428
STASH429
STASH430
STASH431
STASH432
STASH433
STASH434

```



```

      DO 99 J=1,ITERMX
      U=DETK*L*Z0/(K11K21*Z0+2*K12K22*L**2)
      Z1=(NN*PI)+ATAN(U)
      IF(ABS(Z1-Z0).LT.LIMIT) GO TO 9
      IF(BOUND.EQ.(Z1-Z0)) GO TO 90
      BOUND=Z1-Z0
      Z0=Z1
      IF(JPRINT(3).NE.0) WRITE(6,6310),I,J,U,Z1
99   CONTINUE
      WRITE(6,11) I,ITERMX
      STOP
9    EIGEN(I)=Z1
      GO TO 999
90   EIGEN(I)=(Z1+Z0)/2.0
999  CONTINUE
      RETURN

C
C          ERROR MESSAGES
C
11   FORMAT(10X,12HROOT NUMBER ,I3,22H DID NOT CONVERGE AFTER,I5,11H ITERATIONS)
      END
$#
S1BFTC SOLV3  #94/2,XR7
      SUBROUTINE SOLVE3(DETK,EIGEN,NTERMS)
C
C          SOLVES TAN(Z)=0.0
C
      DIMENSION EIGEN(NTERMS)
      DIMENSION JPRINT(3)
      DIMENSION NTAB(5)
      REAL K11K21,K12K22,L,LIMIT,NN
      REAL LIMIT,L,K,LAMBDA,NUMXSS,NUMSS,KFPHI,KFPHIO,KFPHIL,K12
      L,K2C,K22,A,KTERM1,KTERM2
      COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPHUL,CBNX,SNX,DELTA
      IX,DELTAT,K11,K12,K21,K22,FX,NTAB
      COMMON/ROOTS/NTERMS,L,ITERMX,K11K21,K12K22,LIMIT
      COMMON/PRINT/JPRINT
6300  FORMAT(1H1,48X,24HSOLUTION FOR EIGENVALUES//26X,7H ROOT NO.,49X,1H Z)STASH473
6320  FORMAT(2BX,I3,44X,E15.8)
      DATA PI/3.1415926/
      IF(JPRINT(3).NE.0) WRITE(6,6300)
      DO 999 I=1,NTERMS
      NN=I
      Z1=NN*PI
      EIGEN(I)=Z1
      IF(JPRINT(3).NE.0) WRITE(6,6320) I,Z1
999  CONTINUE
      RETURN
      END

$#
S1BFTC SOLV4  #94/2,XR7
      SUBROUTINE SOLVE4(DETK,EIGEN,NTERMS)
C
C          SOLVES TAN(Z)= INFINITY
C
      DIMENSION EIGEN(NTERMS)
      DIMENSION NTAB(5)
      DIMENSION JPRINT(3)
      REAL K11K21,K12K22,L,LIMIT,NN
      REAL LIMIT,L,K,LAMBDA,NUMXSS,NUMSS,KFPHI,KFPHIO,KFPHIL,K12
      K21,K22,N,KTERM1,KTERM2
      STASH435
      STASH436
      STASH437
      STASH438
      STASH439
      STASH440
      STASH441
      STASH442
      STASH443
      STASH444
      STASH445
      STASH446
      STASH447
      STASH448
      STASH449
      STASH450
      STASH451
      STASH452
      STASH453
      STASH454
      STASH455
      STASH456
      STASH457
      STASH458
      STASH459
      STASH460
      STASH461
      STASH462
      STASH463
      STASH464
      STASH465
      STASH466
      STASH467
      STASH468
      STASH469
      STASH470
      STASH471
      STASH472
      STASH473
      STASH474
      STASH475
      STASH476
      STASH477
      STASH478
      STASH479
      STASH480
      STASH481
      STASH482
      STASH483
      STASH484
      STASH485
      STASH486
      STASH487
      STASH488
      STASH489
      STASH490
      STASH491
      STASH492
      STASH493
      STASH494
      STASH495
      STASH496

```

```

COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPUL,CBNX,SBNX,DELTASTASH497
1X,DELTAT,K11,K12,K21,K22,FX,NTAB STASH498
COMMON/ROCTS/NTERPS,L,IYERHA,K1IK2I,K12K22,LIMIT STASH499
COMMON/PRINT/JPRINT STASH500
0300 FORMAT(1H1,46X,24F,SOLUTION FOR EIGENVALUES//26X,7HROOT NO,49X,1H) STASH501
6320 FORMAT(28X,13,44X,E15.8) STASH502
DATA PI/3.1415926/ STASH503
IF(JPRINT(3).NE.0) WRITE(6,6300) STASH504
6300 999 I=1,NTERMS STASH505
NN=I STASH506
ZI=(I2.+NN-1.J+P.I)/2.0 STASH507
EIGEN(I)=ZI STASH508
IF(JPRINT(3).NE.0) WRITE(6,6320) I,ZI STASH509
999 CONTINUE STASH510
RETURN STASH511
END STASH512
59 SIBFTC SIMPS P94/2,XR7 STASH513
REAL FUNCTION FINT(A,B,NN,F) STASH514
C STASH515
C SIMPSON METHOD STASH516
C A=LOWER BOUND STASH517
C B=UPPER BOUND STASH518
C NN=NUMBER OF INTERVALS, MUST BE EVEN STASH519
C F=FUNCTION TO BE INTEGRATED, FUNCTION NAME MUST BE DECLARED STASH520
C EXTERNAL STASH521
C STASH522
C STASH523
REAL LIMIT,L,X,LAMBDA,NUMXSS,NUMSS,KFFPHI,KFFPHIO,KFFPHIL,K12,K12
1,K21,K22,N,KTERM1,KTERM2 STASH524
COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPUL,CBNX,SBNX,DELTASTASH526
XX,DELTAT,K11,K12,K21,K22,FX,NTAB STASH527
FN=NN STASH528
H=(B-A)/NN STASH529
SUMA=0.0 STASH530
SUMB=0.0 STASH531
I=NN-1 STASH532
CO10J=1,H,2 STASH533
FJ=J STASH534
XX=FJ*H STASH535
10 SUMA=SUMA+F(XX) STASH536
I=NN-2 STASH537
CO20KK=2,I,2 STASH538
FK=KK STASH539
XX=FK*H STASH540
SUMB=SUMB+F(XX) STASH541
20 CONTINUE STASH542
FINT=(H/3.*1*(F(A)+F(B))+4.*SUMA+2.*SUMB) STASH543
RETURN STASH544
END STASH545
59 SIBFTC FUNCX P94/2,XR7 STASH546
FUNCTION XFUNC(XX) STASH547
C STASH548
C THIS FUNCTION SETS UP THE INTEGRAND FOR THE / INTEGRAL STASH549
C STASH550
C DIMENSION NTAB(5) STASH551
REAL LIMIT,L,X,LAMBDA,NUMXSS,NUMSS,KFFPHI,KFFPHIO,KFFPHIL,K11,K12
1,K22,N,KTERM1,KTERM2 STASH552
COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPUL,CBNX,SBNX,DELTASTASH555
IX,DELTAT,K11,K12,K21,K22,FX,NTAB STASH556
BNXX=BETAN*XX STASH557
XFUNC=EXPUL*(K12+SIN(BNXX))-K11*BETAN+COS(BNXX)*FX=FX*FCFX(XX) STASH558

```

```

      RETURN
      END
5*
S10FTC FUNCT #94/2,XRT
      FUNCTICH TFUNCTION(LAMBDA)
C
C   THIS FUNCTION SETS UP THE INTEGRAND FOR THE LAMBDA INTEGRAL
C
      REAL LIMIT,L,K,LAP8DA,NUXSS,XUMSS,KFPHI,KFPDIO,KFPHIL,K11,K12
      1,K21,K22,X,XTERM1,KTERM2
      COMMON T,X,LAP8DA,XTERM1,KTERM2,AEPHA,BETAN,EXPML,CBNX,SBNX,DELTASTASH569
      1X,DELTAT,K11,K12,K21,K22,FX,NTAB
      TERM1=XTERM1*PFIPRL(LAMBDA)
      TERM2=XTERM2*PFIPRO(LAMBDA)
      TFUNCTION=TERM1+TERM2*EXP((ALPHA*(BETAN)**2)*(LAMBDA-T))
      RETURN
      END
5*
S10FTC TABLE #94/2,XRT
      SUBROUTINE TABIN(I)
C
C   READS IN TABULAR DATA
C
      1 FORMAT(15.2F10.0+5H
      2      )
      3 FORMAT(1HA,6IX,9HTABLE NO.,I2)
      4 FORMAT(1HD,5DX,34HINDPENDENT
      *     VARIABLE)
      5 FORMAT(4BX,E15.8,1GX,E15.8)
      REAL INOVAR
      DIMENSICK NTAB(5)
      DIMENSION NTABLE(5),INOVAR(5,50),DEPV(45,50)
      COMMON/PTCATA/ INOVAR,DEPVAR
      READ(5,1) NTABLE(1)
      WRITE(6,2) NTABLE(1)
      WRITE(6,3)
      DO 150 J=1,50
      READ(5,13N,1) DEPVARI(1,J),DEPVARI(2,J)
      IF(NTAB(1,3)= 50,100,110
      110 WRITE(6,4) NTAB(1,3),DEPVARI(1,J)
      150 CONTINUE
      WRITE(6,5) NTABLE(1)
      100 RETURN
      50 WRITE(6,6)
      STOP
C
C   ERROR MESSAGES
C
      5 FORMAT(1A,4IX,9HTABLE NO.,I2+29H CONTAINS MORE THAN 50 POINTS)
      5 FORMAT(1A,4BX,27HERROR IN TABULAR INPUT DATA)
      END
6*
S10FTC ENTERP #94/2,XRT
      REAL FUNCTION INTERP(TX,I)
C
C   LINEAR INTERPOLATION FOR TABULAR DATA
C
      DIMENSION NTAB(5)
      DIMENSION INOVAR(5,50),DEPVAR(5,50)
      COMMON/PTCATA/ INOVAR,DEPVAR
      REAL INOVAR
      DO 10 J=1,50
      STASH559
      STASH560
      STASH561
      STASH562
      STASH563
      STASH564
      STASH565
      STASH566
      STASH567
      STASH568
      STASH570
      STASH571
      STASH572
      STASH573
      STASH574
      STASH575
      STASH576
      STASH577
      STASH578
      STASH579
      STASH580
      STASH581
      STASH582
      STASH583
      STASH584
      STASH585
      STASH586
      STASH587
      STASH588
      STASH589
      STASH590
      STASH591
      STASH592
      STASH593
      STASH594
      STASH595
      STASH596
      STASH597
      STASH598
      STASH599
      STASH600
      STASH601
      STASH602
      STASH603
      STASH604
      STASH605
      STASH606
      STASH607
      STASH608
      STASH609
      STASH610
      STASH611
      STASH612
      STASH613
      STASH614
      STASH615
      STASH616
      STASH617
      STASH618
      STASH619
      STASH620

```

```

10 IF(TX-INDVAR(L,J)) 20,30,10 STASH621
CONTINUE STASH622
WRITE(6,99) I STASH623
CALL EXIT STASH624
20 INTERP=DEPVAR(I,J-1)+(DEPVAR(I,J)-DEPVAR(I,J-1))/(TX-INDVAR(I,J-1)) STASH625
=1/(INDVAR(I,J)-INDVAR(I,J-1)) STASH626
GO TO 100 STASH627
30 INTERP=DEPVAR(I,J) STASH628
100 RETURN STASH629
STASH630
C
C          ERROR MESSAGES
C
99 FORMAT(10X,36HARGUMENT EXCEEDS EXTENT OF TABLE NO.,I2) STASH631
END STASH632
$#
$IBFTC PHIOIT N94/2,XR7 STASH633
FUNCTION PHIO(YY) STASH634
STASH635
C
C          THIS FUNCTION CALCULATES THE INSTANTANEOUS VALUE OF THE STASH636
TIME-VARYING BOUNDARY CONDITION AT X=0. STASH637
THE FUNCTION PHIO(YY) MAY BE LOADED INTO THE PROGRAM STASH638
AS AN ANALYTICAL EXPRESSION OR AS POINT DATA IN STASH639
TABULAR FORM STASH640
REAL INTERP STASH641
REAL LIMIT,L,K,LAMBDA,NUMXSS,NUMSS,KFPHI,KFPHIO,KFPHIL,K11,K12 SYASH642
1,K21,K22,N,KTERM1,KTERM2 STASH643
COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPML,CBNX,SBNX,DELTA STASH644
IX,DELTAT,K11,K12,K21,K22,FX,NTAB STASH645
DIMENSION NTAB(5) STASH646
IF(NTAB(1).NE.0) GO TO 100 STASH647
C
C          PHIO=ANY FUNCTION OF TIME STASH648
C
C          PHIO=1.0 STASH649
RETURN STASH650
100 PHIO=INTERP(YY,1) STASH651
RETURN STASH652
END STASH653
STASH654
$#
$IBFTC PHILT N94/2,XR7 STASH655
FUNCTION PHIL(YY) STASH656
STASH657
C
C          THIS FUNCTION CALCULATES THE INSTANTANEOUS VALUE OF THE STASH658
TIME-VARYING BOUNDARY CONDITION AT X=L. STASH659
THE FUNCTION PHIL(YY) MAY BE LOADED INTO THE PROGRAM STASH660
AS AN ANALYTICAL EXPRESSION OR AS POINT DATA IN STASH661
TABULAR FORM. STASH662
REAL INTERP STASH663
REAL LIMIT,L,K,LAMBDA,NUMXSS,NUMSS,KFPHI,KFPHIO,KFPHIL,K11,K12 SYASH664
1,K21,K22,N,KTERM1,KTERM2 STASH665
COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPML,CBNX,SBNX,DELTA STASH666
IX,DELTAT,K11,K12,K21,K22,FX,NTAB STASH667
DIMENSION NTAB(5) STASH668
IF(NTAB(1).NE.0) GO TO 100 STASH669
C
C          PHIL=ANY FUNCTION OF YY STASH670
C
C          PHIL=1.0 STASH671
RETURN STASH672
100 PHIL=INTERP(YY,2) STASH673
STASH674
STASH675
STASH676
STASH677
STASH678
STASH679
STASH680
STASH681
STASH682

```

```

      RETURN
      END
%
$1BFTC DERIVO #94/2,XR7
      FUNCTIONNPHPRO(LAMBDA)
C
C THIS FUNCTION CALCULATES THE INSTANTANEOUS VALUE OF THE
C DERIVATIVE OF THE TIME-VARYING BOUNDARY CONDITION AT X=0.
C THE FUNCTION MAY BE LOADED ANALYTICALLY OR AS POINT
C DATA IN TABULAR FORM.
C
      REAL INTERP
      REAL LIMIT,L,K,LAMBDA,NUMXSS,NUNSS,KFPHI,KFPHIO,KFPHIL,K11,K12
      1,K21,K22,N,KTERM1,KTERM2
      COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPHUL,CBNX,SBNX,DELTASTASH69
      1X,DELTAT,K11,K12,K21,K22,FX,NTAB
      DIMENSION NTAB(5)
      IF(NTAB(3).NE.0) GO TO 100
C
C PHIPRO=ANY FUNCTION OF TIME
C
      PHIPRO=0.
      RETURN
100  PHIPRO=INTERP(LAMBDA,3)
      RETURN
      END
%
$1BFTC DERIVL #94/2,XR7
      FUNCTIONNPHPRL(LAMBDA)
C
C THIS FUNCTION CALCULATES THE INSTANTANEOUS VALUE OF THE
C DERIVATIVE OF THE TIME-VARYING BOUNDARY CONDITIONS AT X=L.
C THE FUNCTION MAY BE LOADED ANALYTICALLY OR AS POINT DATA
C IN TABULAR FORM
C
      REAL INTERP
      REAL LIMIT,L,K,LAMBDA,NUMXSS,NUNSS,KFPHI,KFPHIO,KFPHIL,K11,K12
      1,K21,K22,N,KTERM1,KTERM2
      COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPHUL,CBNX,SBNX,DELTASTASH71
      1X,DELTAT,K11,K12,K21,K22,FX,NTAB
      DIMENSION NTAB(5)
      IF(NTAB(4).NE.0) GO TO 100
C
C PHIPRL=ANY FUNCTION OF TIME
C
      PHIPRL=0.
      RETURN
100  PHIPRL=INTERP(LAMBDA,4)
      RETURN
      END
%
$1BFTC FX #94/2+XR7
      FUNCTIONNFCFXL()
C
C THIS FUNCTION COMPUTES THE INITIAL CONDITIONS OF THE ROD.
C THESE INITIAL CONDITIONS MAY BE LOADED INTO THE PROGRAM
C ANALYTICALLY OR AS POINT DATA IN TABULAR FORM.
C
      REAL INTERP
      REAL LIMIT,L,K,LAMBDA,NUMXSS,NUNSS,KFPHI,KFPHIO,KFPHIL,K11,K12
      1,K21,K22,N,KTERM1,KTERM2
      COMMON T,X,LAMBDA,KTERM1,KTERM2,ALPHA,BETAN,EXPHUL,CBNX,SBNX,DELTASTASH74

```

```
1X,DELTAT,X11,X12,X21,X22,FX,4768          STASH745
      DIMENSION NYAB(5)                      STASH746
      IF(NYAB(5).NE.0) GO TO 100                STASH747
C
C      F0FX= ANY FUNCTION OF X                 STASH748
C
C      F0FX=1.                                 STASH749
55      RETURN                                STASH750
100     F0FX=INTERP(XX,5)                     STASH751
      RETURN                                STASH752
      END                                    STASH753
                                         STASH754
                                         STASH755
```

APPENDIX II  
EIGENVALUE SUBROUTINES

APPENDIX II  
EIGENVALUE SUBROUTINES

The solution of equation 53 depends on values of  $\beta_n$  and which are derived from the positive eigenvalues of Equation 40.

$$\tan z_n = \frac{D L Z_n}{K_{21} K_{11} Z_n^2 + K_{22} K_{12}} \frac{L^2}{L^2} \quad (40)$$

Since we are seeking positive values of  $z_n$  the sign of the left-hand side of the equation may be associated with the parameter, D. Thus three formulations are possible corresponding to D being positive, negative or zero. A fourth possibility is that of the denominator going to zero. The last two solutions are trivial. If D is zero we have

$$\tan z_n = 0$$

The solution to this equation is merely

$$z_n = n\pi \quad (54)$$

If the denominator of equation 40 goes to zero we have

$$\tan z_n = \infty$$

The solution to the equation is

$$z_n = \frac{(2n-1)\pi}{2} \quad (55)$$

However, if D has a value other than zero the equations are solved by an iterative process. In these cases the eigenvalue subroutine has been programmed to ignore the root at (0,0) since this produces a trivial solution. The procedure then is outlined below for a positive value of D.

We know that the solution lies between

$$\left[ (n-1)\pi, \frac{(2n-1)\pi}{2} \right], \quad n = 1, 2, 3, \dots$$

For a first approximation to the root,  $z_{n,0}$ , we shall choose

$$z_{n,0} = \frac{(2n-1)\pi}{2} \quad (56)$$

We then write two equations

$$u_{n,m} = \frac{D L Z_{n,m-1}}{K_{21} K_{11} Z_{n,m-1}^2 + K_{22} K_{12}} \frac{L^2}{L^2} \quad (57)$$

$$Z_{n,m} = (n-1)\pi + \tan^{-1}(u_{n,m}) \quad (58)$$

Where the subscripts n and m refer to the m<sup>th</sup> iteration toward the n<sup>th</sup> root. The root is then determined to any desired LIMIT of accuracy by writing

$$|Z_{n,m} - Z_{n,m-1}| < \text{LIMIT} \quad (59)$$

Provision is made in the program to print out the iteration steps should any trouble occur. The basic difference in the solution for a negative value of D arises in the first approximation,  $Z_{n,0}$ . For negative values of D Equation 56 becomes

$$Z_{n,0} = (n + 1) \pi \quad (60)$$

and the solution proceeds as before with Equation 58 becoming

$$Z_{n,m} = n\pi + \tan^{-1}(u_{n,m}) \quad (61)$$

AFFDL-TR-66-109

PREVIOUS PAGE WAS BLANK, THIS PAGE IS NOT PAGED.

APPENDIX III  
RESULTS OF CHECK PROBLEMS

TABLE II  
COMPARISON OF DATA FOR CASE I

| L    | $T(0,t)=t; T(L,t)=0; T(x,0)=0$ |                  |                   |                    |        |         |         |         |
|------|--------------------------------|------------------|-------------------|--------------------|--------|---------|---------|---------|
|      | TEMP @<br>$t=0$                | TEMP @<br>$t=50$ | TEMP @<br>$t=500$ | TEMP @<br>$t=1000$ | STASH  | LTA     | STASH   | LTA     |
| 0.0  | 0.00                           | 50.00            | 50.00             | 500.00             | 500.00 | 1000.00 | 1000.00 | 1000.00 |
| 0.5  | 0.00                           | 37.39            | 37.38             | 455.95             | 455.93 | 930.71  | 930.69  | 930.69  |
| 1.0  |                                | 27.45            | 27.46             | 414.85             | 414.83 | 864.38  | 864.35  | 864.35  |
| 1.5  |                                | 19.79            | 19.80             | 376.56             | 376.53 | 800.87  | 800.83  | 800.83  |
| 2.0  |                                | 13.99            | 14.00             | 340.92             | 340.87 | 740.02  | 739.97  | 739.97  |
| 2.5  |                                | 9.76             | 9.70              | 307.74             | 307.69 | 681.66  | 681.60  | 681.60  |
| 3.0  |                                | 6.58             | 6.59              | 276.88             | 276.83 | 625.65  | 625.59  | 625.59  |
| 3.5  |                                | 4.37             | 4.38              | 248.18             | 248.12 | 571.82  | 571.76  | 571.76  |
| 4.0  |                                | 2.84             | 2.85              | 221.48             | 221.42 | 520.63  | 519.96  | 519.96  |
| 4.5  |                                | 1.80             | 1.81              | 196.61             | 196.56 | 470.11  | 470.04  | 470.04  |
| 5.0  |                                | 1.12             | 1.13              | 173.43             | 173.37 | 421.90  | 421.84  | 421.84  |
| 5.5  |                                | 0.68             | 0.68              | 151.77             | 151.72 | 375.26  | 375.20  | 375.20  |
| 6.0  |                                | 0.40             | 0.41              | 131.48             | 131.43 | 330.03  | 329.97  | 329.97  |
| 6.5  |                                | 0.23             | 0.24              | 112.49             | 112.36 | 286.04  | 285.99  | 285.99  |
| 7.0  |                                | 0.13             | 0.13              | 94.38              | 94.34  | 243.15  | 243.10  | 243.10  |
| 7.5  |                                | 0.07             | 0.07              | 77.27              | 77.24  | 201.19  | 201.15  | 201.15  |
| 8.0  |                                | 0.04             | 0.04              | 60.91              | 60.89  | 160.02  | 159.98  | 159.98  |
| 8.5  |                                | 0.02             | 0.02              | 45.16              | 45.14  | 119.47  | 119.44  | 119.44  |
| 9.0  |                                | 0.01             | 0.01              | 29.86              | 29.84  | 79.38   | 79.37   | 79.37   |
| 9.5  |                                | 0.00             | 0.00              | 14.85              | 14.84  | 39.61   | 39.60   | 39.60   |
| 10.0 | 0.00                           | 0.00             | 0.00              | 0.00               | 0.00   | 0.00    | 0.00    | 0.00    |

TABLE III  
COMPARISON OF DATA FOR CASE 2

| L    | $T(0,t)=100+t; T(L,t)=10^2; T(x,0)=10^2$ |                |                 |                  |        |        |         |         |
|------|--|----------------|-----------------|------------------|--------|--------|---------|---------|
|      | TEMP @<br>t=0                            | TEMP @<br>t=50 | TEMP @<br>t=500 | TEMP @<br>t=1000 | STASH  | LTA    | STASH   | LTA     |
| 0.0  | 100.00                                   | 150.00         | 600.00          | 1100.00          | 150.00 | 150.00 | 1100.00 | 1100.00 |
| 0.5  | 100.00                                   | 137.38         | 555.95          | 1030.71          | 137.38 | 137.38 | 1030.69 | 1030.69 |
| 1.0  |  | 127.45         | 514.85          | 964.38           | 127.45 | 127.46 | 964.35  | 964.35  |
| 1.5  |  | 119.79         | 476.56          | 900.87           | 119.79 | 119.80 | 900.83  | 900.83  |
| 2.0  |  | 113.99         | 440.92          | 840.02           | 113.99 | 114.00 | 839.97  | 839.97  |
| 2.5  |  | 109.70         | 407.74          | 781.66           | 109.70 | 109.70 | 781.66  | 781.66  |
| 3.0  |  | 106.58         | 376.88          | 725.59           | 106.58 | 106.59 | 725.59  | 725.59  |
| 3.5  |  | 104.37         | 348.18          | 671.76           | 104.37 | 104.38 | 671.76  | 671.76  |
| 4.0  |  | 102.84         | 321.48          | 619.96           | 102.84 | 102.85 | 619.96  | 619.96  |
| 4.5  |  | 101.80         | 296.61          | 570.04           | 101.80 | 101.81 | 570.04  | 570.04  |
| 5.0  |  | 101.11         | 273.43          | 521.84           | 101.11 | 101.13 | 521.84  | 521.84  |
| 5.5  |  | 100.68         | 251.77          | 475.20           | 100.68 | 100.68 | 475.20  | 475.20  |
| 6.0  |  | 100.40         | 231.48          | 429.97           | 100.40 | 100.40 | 429.97  | 429.97  |
| 6.5  |  | 100.23         | 212.40          | 385.99           | 100.23 | 100.24 | 385.99  | 385.99  |
| 7.0  |  | 100.13         | 194.38          | 343.10           | 100.13 | 100.13 | 343.10  | 343.10  |
| 7.5  |  | 100.07         | 177.27          | 301.15           | 100.07 | 100.07 | 301.15  | 301.15  |
| 8.0  |  | 100.04         | 160.91          | 259.98           | 100.04 | 100.04 | 259.98  | 259.98  |
| 8.5  |  | 100.02         | 145.16          | 219.44           | 100.02 | 100.02 | 219.44  | 219.44  |
| 9.0  |  | 100.01         | 129.36          | 179.37           | 100.01 | 100.01 | 179.37  | 179.37  |
| 9.5  |  | 100.00         | 114.85          | 139.60           | 100.00 | 100.00 | 139.60  | 139.60  |
| 10.0 | 100.00                                   | 100.00         | 100.00          | 100.00           | 100.00 | 100.00 | 100.00  | 100.00  |

TABLE IV  
COMPARISON OF DATA FOR CASE 3

| L    | TEMP @<br>t=0 | TEMP @ t=50 |        | TEMP @ t=500 |        | TEMP @ t=1000 |         |
|------|---------------|-------------|--------|--------------|--------|---------------|---------|
|      |               | STASH       | LTA    | STASH        | LTA    | STASH         | LTA     |
| 0.0  | 0.00          | 50.00       | 50.00  | 500.00       | 500.00 | 1000.00       | 1000.00 |
| 0.5  | 5.00          | 42.38       | 42.38  | 460.95       | 460.93 | 935.71        | 935.69  |
| 1.0  | 10.00         | 37.46       | 37.46  | 424.85       | 424.83 | 874.38        | 874.35  |
| 1.5  | 15.00         | 34.79       | 34.80  | 391.56       | 391.53 | 815.87        | 815.83  |
| 2.0  | 20.00         | 33.99       | 34.00  | 360.92       | 360.87 | 760.02        | 759.97  |
| 2.5  | 25.00         | 34.70       | 34.70  | 322.74       | 332.69 | 706.66        | 706.60  |
| 3.0  | 30.00         | 36.58       | 36.59  | 306.88       | 306.83 | 655.65        | 655.59  |
| 3.5  | 35.00         | 39.37       | 39.38  | 283.18       | 283.16 | 606.82        | 606.76  |
| 4.0  | 40.00         | 42.84       | 42.85  | 261.48       | 261.42 | 560.03        | 559.96  |
| 4.5  | 45.00         | 46.80       | 46.81  | 241.61       | 241.56 | 515.11        | 515.04  |
| 5.0  | 50.00         | 51.11       | 51.13  | 223.43       | 223.37 | 471.90        | 471.84  |
| 5.5  | 55.00         | 55.68       | 55.68  | 206.77       | 206.72 | 430.26        | 430.20  |
| 6.0  | 60.00         | 60.40       | 60.41  | 191.48       | 191.43 | 390.03        | 389.97  |
| 6.5  | 65.00         | 65.23       | 65.24  | 177.40       | 177.36 | 351.04        | 350.99  |
| 7.0  | 70.00         | 70.13       | 70.13  | 164.38       | 164.34 | 313.15        | 313.10  |
| 7.5  | 75.00         | 75.07       | 75.07  | 152.27       | 152.24 | 276.19        | 276.15  |
| 8.0  | 80.00         | 80.04       | 80.04  | 140.91       | 140.89 | 240.02        | 239.98  |
| 8.5  | 85.00         | 85.02       | 85.02  | 130.16       | 130.14 | 204.47        | 204.44  |
| 9.0  | 90.00         | 90.01       | 90.01  | 119.86       | 119.84 | 169.38        | 169.37  |
| 9.5  | 95.00         | 95.00       | 95.00  | 109.85       | 109.84 | 134.61        | 134.60  |
| 10.0 | 100.00        | 100.00      | 100.00 | 100.00       | 100.00 | 100.00        | 100.00  |

## COMPARISON OF ALGORITHMS FOR CASE 3

| L    | TEMP @ t=0 |        | TEMP @ t=50 |        | TEMP @ t=500 |        | TEMP @ t=1000 |        |
|------|------------|--------|-------------|--------|--------------|--------|---------------|--------|
|      | STASH      | LTA    | STASH       | LTA    | STASH        | LTA    | STASH         | LTA    |
| 0.0  | 0.00       | 50.00  | 50.00       | 500.00 | 360.00       | 360.00 | 360.00        | 360.00 |
| 0.5  | 5.00       | 42.38  | 42.38       | 460.95 | 460.95       | 935.71 | 935.69        | 935.69 |
| 1.0  | 10.00      | 37.46  | 37.46       | 434.85 | 424.83       | 811.35 | 811.35        | 811.35 |
| 1.5  | 15.00      | 34.79  | 34.80       | 391.56 | 391.53       | 811.37 | 813.83        | 813.83 |
| 2.0  | 20.00      | 33.99  | 34.00       | 360.92 | 360.87       | 760.02 | 762.97        | 762.97 |
| 2.5  | 25.00      | 34.70  | 34.70       | 332.74 | 332.69       | 706.66 | 708.60        | 708.60 |
| 3.0  | 30.00      | 36.58  | 36.59       | 306.88 | 306.83       | 655.65 | 655.59        | 655.59 |
| 3.5  | 35.00      | 39.37  | 39.38       | 283.18 | 283.16       | 606.82 | 606.76        | 606.76 |
| 4.0  | 40.00      | 42.84  | 42.85       | 261.48 | 261.42       | 560.03 | 559.96        | 559.96 |
| 4.5  | 45.00      | 46.80  | 46.81       | 241.61 | 241.56       | 515.11 | 515.04        | 515.04 |
| 5.0  | 50.00      | 51.11  | 51.13       | 223.43 | 223.37       | 471.90 | 471.84        | 471.84 |
| 5.5  | 55.00      | 55.68  | 55.68       | 206.77 | 206.72       | 430.26 | 430.20        | 430.20 |
| 6.0  | 60.00      | 60.49  | 60.41       | 191.48 | 191.43       | 390.03 | 389.97        | 389.97 |
| 6.5  | 65.00      | 65.23  | 65.24       | 177.40 | 177.36       | 351.04 | 350.99        | 350.99 |
| 7.0  | 70.00      | 70.12  | 70.13       | 164.38 | 164.34       | 313.15 | 313.10        | 313.10 |
| 7.5  | 75.00      | 75.07  | 75.07       | 152.27 | 152.24       | 276.19 | 276.15        | 276.15 |
| 8.0  | 80.00      | 80.04  | 80.04       | 140.91 | 140.89       | 240.02 | 239.98        | 239.98 |
| 8.5  | 85.00      | 85.02  | 85.02       | 130.16 | 130.14       | 204.47 | 204.44        | 204.44 |
| 9.0  | 90.00      | 90.01  | 90.01       | 119.86 | 119.84       | 169.38 | 169.37        | 169.37 |
| 9.5  | 95.00      | 95.00  | 95.00       | 109.85 | 109.84       | 134.61 | 134.60        | 134.60 |
| 10.0 | 100.00     | 100.00 | 100.00      | 100.00 | 100.00       | 100.00 | 100.00        | 100.00 |

TABLE V  
COMPARISON OF DATA FOR CASE 4

| L    | $T(0, t) = 1000; q(L, t) = 0; T(x, 0) = 100$ |         |                |         |                 |         |                 |         |
|------|--|---------|----------------|---------|-----------------|---------|-----------------|---------|
|      | TEMP @ $t=0$                                 |         | TEMP @ $t=300$ |         | TEMP @ $t=1800$ |         | TEMP @ $t=3600$ |         |
|      | STASH  | LTA     | STASH          | LTA     | STASH           | LTA     | STASH           | LTA     |
| 1.0  | 100.00                                       | 1000.00 | 1000.00        | 1000.00 | 1000.00         | 1000.00 | 1000.00         | 1000.00 |
| 2.0  | 100.00                                       | 899.95  | 899.99         | 994.85  | 994.40          | 999.85  | 999.80          | 999.80  |
| 3.0  | 100.00                                       | 802.53  | 802.50         | 989.83  | 989.45          | 999.71  | 999.65          | 999.65  |
| 4.0  | 100.00                                       | 710.26  | 709.99         | 985.06  | 984.72          | 999.57  | 999.53          | 999.53  |
| 5.0  | 100.00                                       | 625.48  | 625.30         | 980.06  | 978.98          | 999.44  | 999.41          | 999.41  |
| 6.0  | 100.00                                       | 550.27  | 550.05         | 976.72  | 976.13          | 999.33  | 999.27          | 999.27  |
| 7.0  | 100.00                                       | 486.38  | 486.14         | 973.37  | 972.66          | 999.24  | 999.19          | 999.19  |
| 8.0  | 100.00                                       | 435.25  | 434.97         | 970.67  | 970.04          | 999.16  | 999.12          | 999.12  |
| 9.0  | 100.00                                       | 397.98  | 397.52         | 968.69  | 968.08          | 999.10  | 999.06          | 999.06  |
| 10.0 | 100.00                                       | 375.51  | 375.02         | 967.49  | 967.10          | 999.07  | 999.01          | 999.01  |
|      | 100.00                                       | 367.70  | 367.56         | 967.08  | 966.99          | 999.05  | 998.99          | 998.99  |

TABLE VI  
COMPARISON OF DATA FOR CASE 5

| L    | $T(0, t) = 0; q(z, t) = \frac{1}{1m^2} \text{sec}^{-1}; T(x, 0) = 0$ |         |                |         |                 |         |                 |         |
|------|--|---------|----------------|---------|-----------------|---------|-----------------|---------|
|      | TEMP @ $t=0$   |         | TEMP @ $t=300$ |         | TEMP @ $t=1800$ |         | TEMP @ $t=3600$ |         |
|      | STASH  | LTA     | STASH          | LTA     | STASH           | LTA     | STASH           | LTA     |
| 0.0  | 0.0  | 0.0     | 0.0            | 0.0     | 0.0             | 0.0     | 0.0             | 0.0     |
| 1.0  | 0.0  | 129.76  | 129.08         | 416.41  | 416.34          | 431.70  | 431.69          | 431.69  |
| 2.0  | 0.0  | 266.80  | 265.45         | 833.21  | 833.08          | 863.41  | 863.39          | 863.39  |
| 3.0  | 0.0  | 418.26  | 416.26         | 1250.77 | 1250.58         | 1295.14 | 1295.12         | 1295.12 |
| 4.0  | 0.0  | 590.98  | 588.36         | 1669.46 | 1669.22         | 1726.91 | 1726.88         | 1726.88 |
| 5.0  | 0.0  | 791.36  | 788.17         | 2089.61 | 2099.32         | 2158.72 | 2158.68         | 2158.68 |
| 6.0  | 0.0  | 1025.20 | 1021.51        | 2511.51 | 2511.17         | 2594.51 | 2594.54         | 2594.54 |
| 7.0  | 0.0  | 1297.54 | 1293.43        | 2934.41 | 2935.04         | 3022.49 | 3022.45         | 3022.45 |
| 8.0  | 0.0  | 1612.46 | 1608.04        | 3361.52 | 3361.12         | 3454.47 | 3454.47         | 3454.47 |
| 9.0  | 0.0  | 1973.03 | 1968.41        | 3789.99 | 3789.57         | 3883.51 | 3886.47         | 3886.47 |
| 10.0 | 0.0  | 2381.11 | 2376.43        | 4220.90 | 4220.48         | 4318.67 | 4318.58         | 4318.58 |

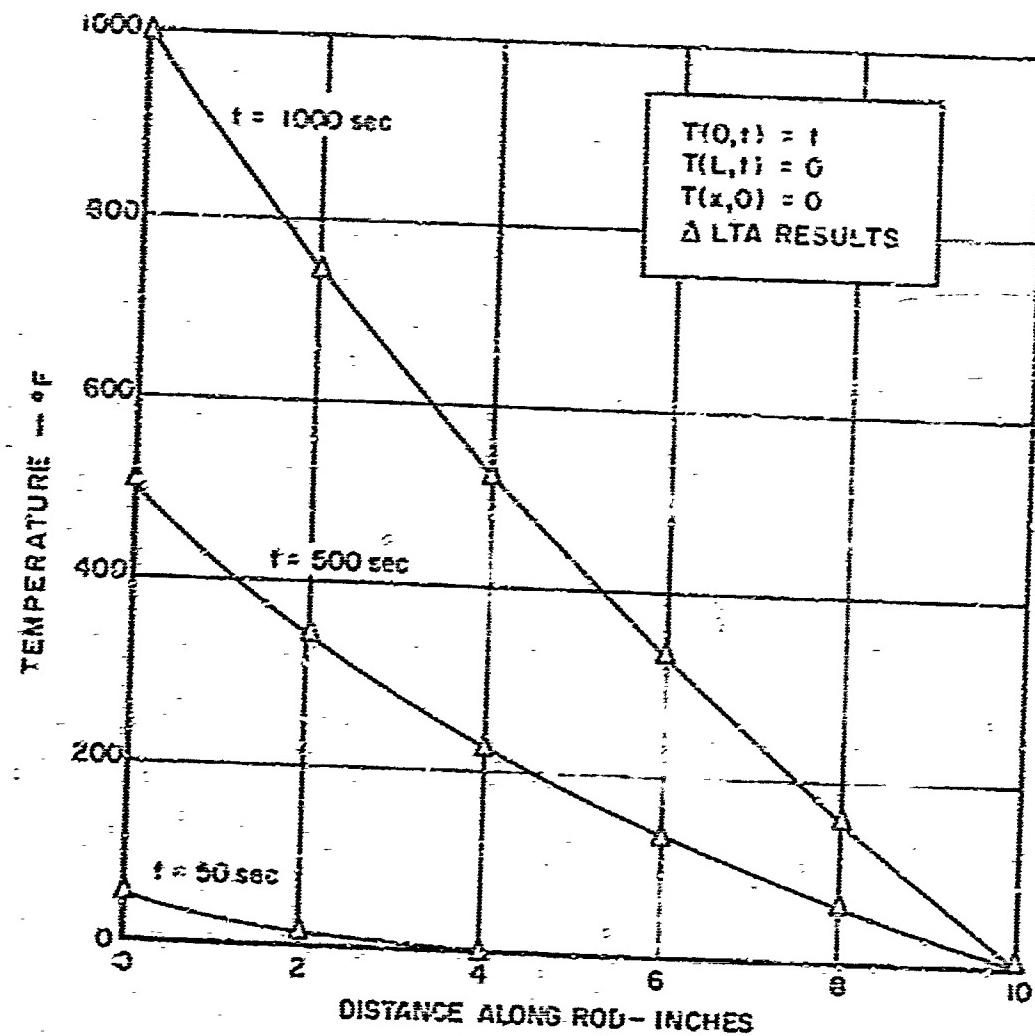


Figure 8. Temperature Profiles (Case 2)

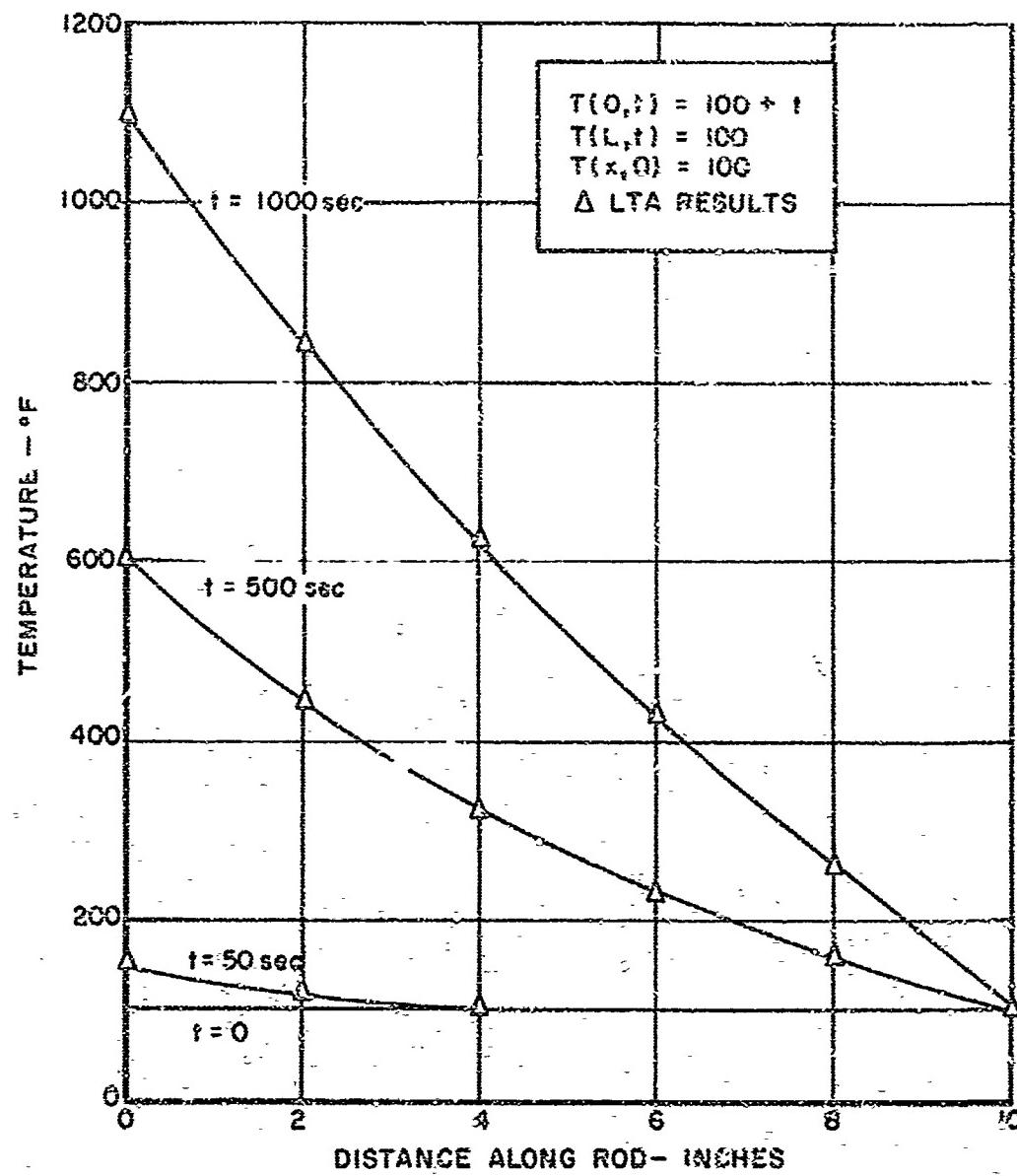


Figure 9. Temperature Profiles (Case 2)

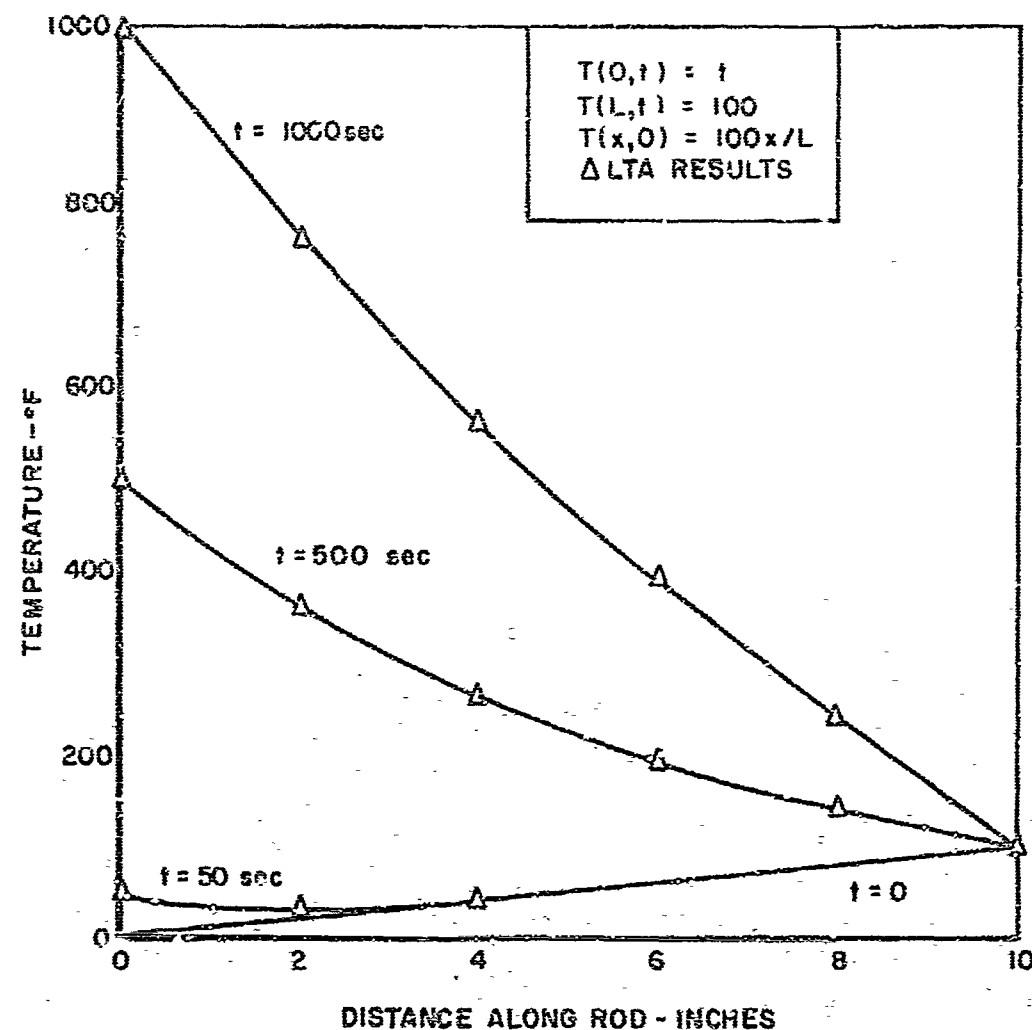


Figure 10. Temperature Profiles (Case 3)

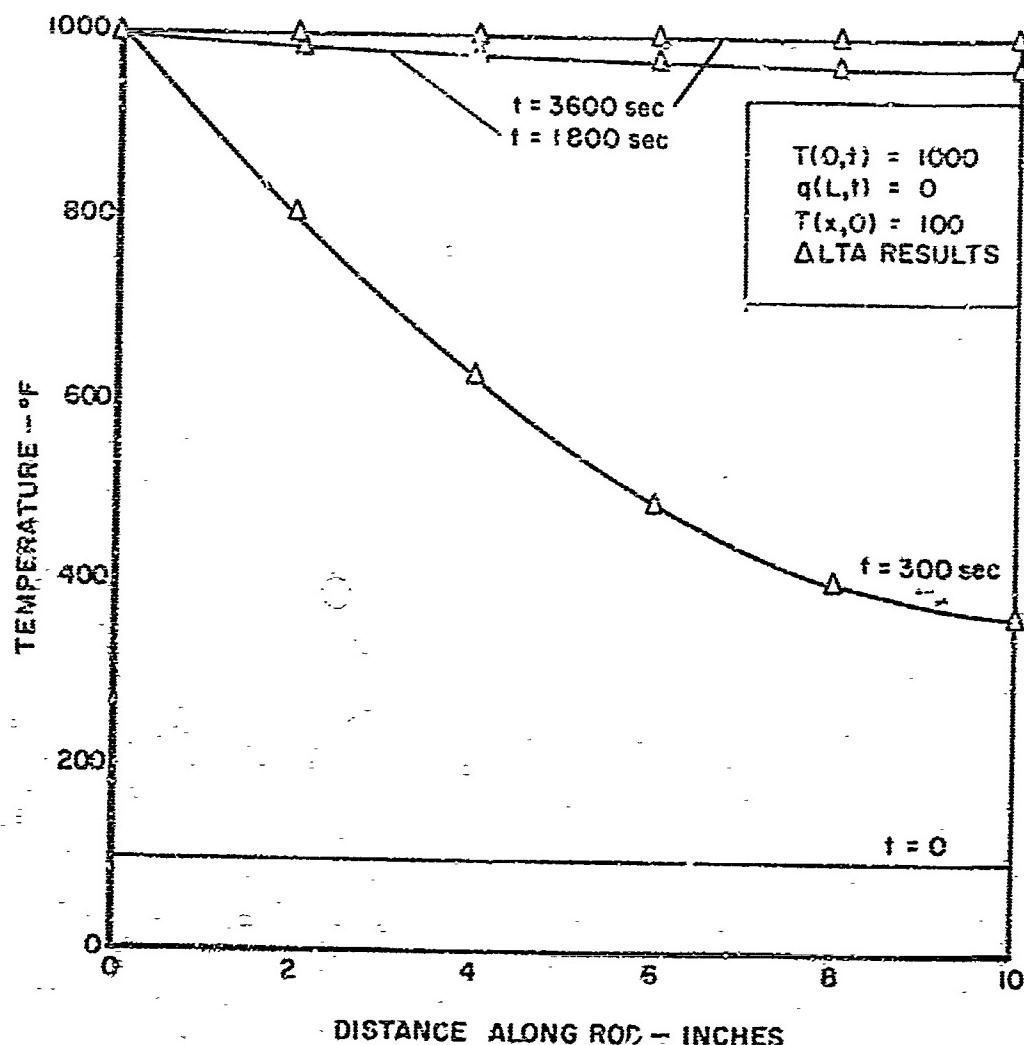


Figure 11. Temperature Profiles (Case 4)

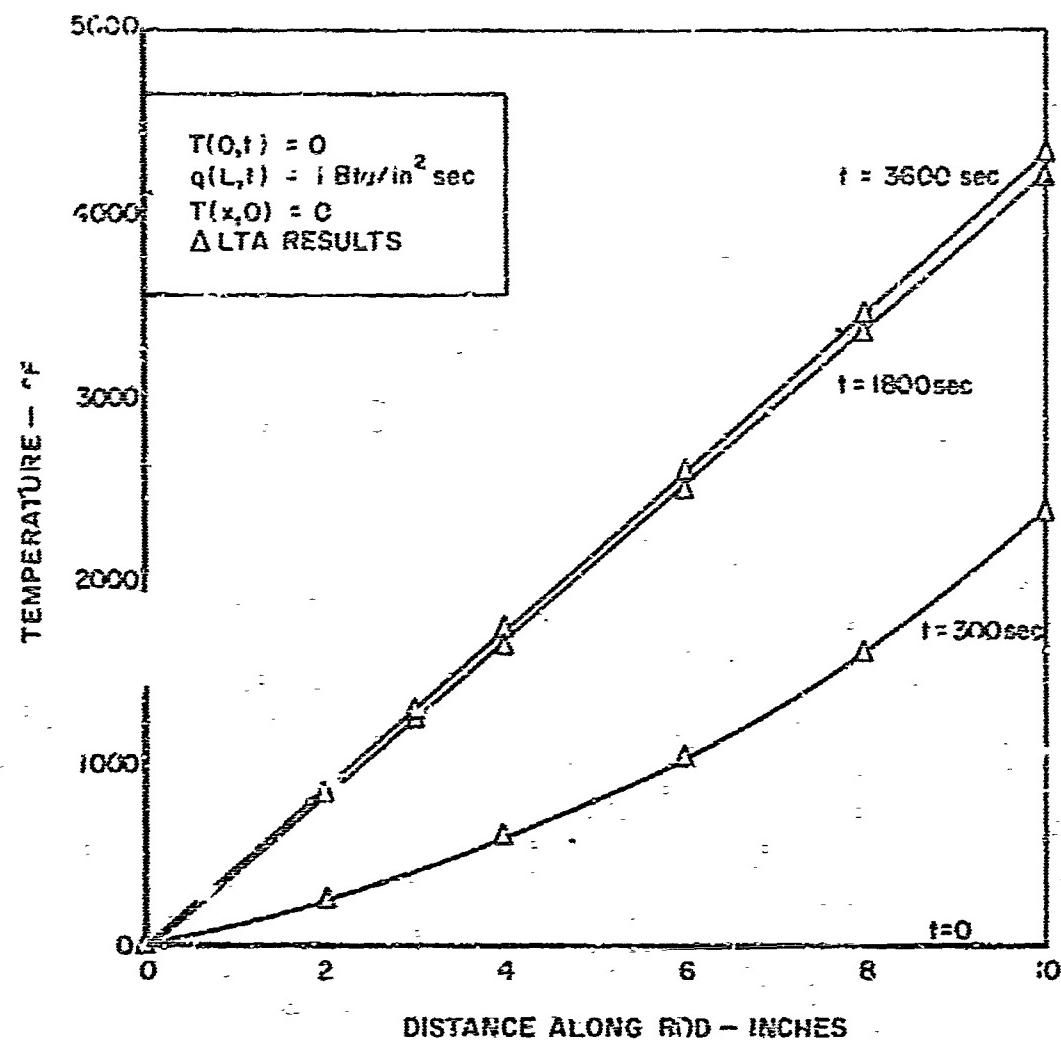


Figure 12. Temperature Profiles (Case 5)

Unclassified

Security Classification

DOCUMENT CONTROL DATA - P&D

(Security classification of title, body of abstract and indexing information must be entered unless the document report is classified)

|   |  |   |  |
|---|--|---|--|
| 1. ORIGINATING ACTIVITY (Corporate Author)  |  | 2. REPORT SECURITY CLASSIFICATION   |  |
| Air Force Flight Dynamics Laboratory, Research and Technology Division, Air Force Systems Command, Wright-Patterson Air Force Base, Ohio 45433  |  | 2A CPOC   |  |
| 3. REPORT TITLE   |  |   |  |
| TRANSIENT ANALYSIS OF HEAT CONDUCTION THROUGH A SLAB BY INFINITE SERIES   |  |   |  |
| 4. DESCRIPTIVE NOTES (Type of report and inclusive dates)   |  |   |  |
| 5. AUTHOR(S) (Last name, first name, middle)  |  |   |  |
| Bernstein, Thomas N.,<br>Engle, Robert M., Jr.  |  |   |  |
| 6. REPORT DATE  | 7A. TOTAL NO. OF PAGES   | 7B. NO. OF REPS   |  |
| December 1966   | 64   |   |  |
| 8A. CONTRACT OR GRANT NO.   | 8B. ORIGINATOR'S REPORT NUMBER(S)  |   |  |
| 9A. PROJECT NO. 1467  | AFFDL-TR-66-103  |   |  |
| 9B. Task No. 146702   | 9C. OTHER REPORT NO(S) (List other numbers that may be assigned this report) |   |  |
| 10. AVAILABILITY/LIMITATION NOTICES This document is subject to special export controls and each transmittal to foreign governments or foreign nationals may be made only with prior approval of the Theoretical Mechanics Branch, Structures Division, Air Force Flight Dynamics Laboratory (FDTR), Wright-Patterson Air Force Base, Ohio 45433  |  |   |  |
| 11. SUPPLEMENTARY NOTES   |  | 12. ACQUISITION MILITARY ACTIVITY   |  |
|   |  | Air Force Flight Dynamics Laboratory,<br>Research and Technology Division, Air Force<br>Systems Command, Wright-Patterson AFB, Ohio |  |
| 13. ABSTRACT  |  |   |  |
| The exact solution to the problem of conduction of heat through a slab is developed. The solution, formulated in terms of an infinite series, allows arbitrary initial conditions and time-dependent boundary conditions. The solution is programmed in FORTRAN IV for the IBM 7034 II computer. Several check problems were solved and the results were compared with those obtained from a finite difference heat transfer program. |  |   |  |

DD FORM 1473

Unclassified  
Security Classification

Unclassified

Sec. 1.4. Classification

| 4<br>KEY WORDS   | LINK A |    | LINK B |    | LINK C |    |
|--|--------|----|--------|----|--------|----|
|  | ROLE   | WT | ROLE   | WT | ROLE   | WT |
| Thermal Analysis<br>Mathematical Formulation<br>Digital Computer Analysis<br>Arbitrary Boundary Conditions<br>Infinite Series<br>Computer Program<br>One-Dimensional<br>Tridiagonal<br>Convection<br>FORTRAN |        |    |        |    |        |    |

INSTRUCTIONS

1. ORIGINATING ACTIVITY: Enter the name and address of the contractor, subcontractor, grants Department of Defense security or other organization (corporate author) issuing the report.

2. REPORT SECURITY CLASSIFICATION: Enter the overall security classification of the report. Indicate whether "Restricted Data" is included. Referring to DOD Directive 5200.3 and Armed Forces Industrial Manual, Enter the group number. Also, where applicable, show that optional markings have been used for Group 3 and Group 4 as authorized.

3. REPORT TITLE: Enter the complete report title in all capital letters. Titles in all cases should be unclassified. If a meaningful title cannot be selected without classification, show title classification in all capital in parentheses immediately following the title.

4. DESCRIPTIVE NOTES: If appropriate, enter the type of report, e.g., interim, progress, summary, annual, or final. Give the inclusive dates when a specific reporting period is covered.

5. AUTHOR(S): Enter the name(s) of author(s) as shown on or in the report. Enter last name, first name, middle initial. If military, show rank and branch of service. The name of the principal author is an absolute minimum requirement.

6. REF. DRT DATE: Enter the date of the report as day, month, year, >month, year. If more than one date appears on the report, use date of publication.

7a. TOTAL NUMBER OF PAGES: The total page count should follow normal pagination procedures, i.e., enter the number of pages containing information.

7b. NUMBER OF REFERENCES: Enter the total number of references cited in the report.

8a. CONTRACT OR GRANT NUMBER: If appropriate, enter the applicable number of the contract or grant under which the report was utilized.

8b, c, & 8d. PROJECT NUMBER: Enter the appropriate project identification no., such as project number, sub-project number, system numbers, task number, etc.

9a. ORIGINATOR'S REPORT NUMBER(S): Enter the official report number by which the document will be identified and controlled by the originating activity. This number must be unique to this report.

9b. OTHER REPORT NUMBER(S): If the report has been assigned any other report numbers (either by the originator or by the sponsor), also enter this number(s).

10. AVAILABILITY/LIMITATION NOTICES: Enter any limitations on further dissemination of the report, other than those

imposed by security classification, using standard statements such as:

- (1) "Qualified requesters may obtain copies of this report from DDC."
- (2) "Foreign announcement and dissemination of this report by DDC is not authorized."
- (3) "U. S. Government agencies may obtain copies of this report directly from DDC. Other qualified DDC users shall request through"
- (4) "... U. S. military agencies may obtain copies of this report directly from DDC. Other qualified users shall request through"
- (5) "All distribution of this report is controlled. Qualified DDC users shall request through"

If the report has been furnished to the Office of Technical Services, Department of Commerce, for sale to the public, indicate this fact and enter the price, if known.

11. SUPPLEMENTARY NOTES: Use for additional explanatory notes.

12. SPONSORING MILITARY ACTIVITY: Enter the name of the departmental project office or laboratory sponsoring (paying for) the research and development. Include address.

13. ABSTRACT: Enter an abstract giving a brief and factual summary of the document indicative of the report, even though it may also appear elsewhere in the body of the technical report. If additional space is required, a continuation sheet shall be attached.

It is highly desirable that the abstract of classified reports be unclassified. Each paragraph of the abstract shall end with an indication of the military security classification of the information in the paragraph, represented as (T), (S), (C), or (U). There is no limitation on the length of the abstract. However, the suggested length is from 150 to 225 words.

14. KEY WORDS: Key words are technically meaningful terms or short phrases that characterize a report and may be used as index entries for cataloging the report. Key words must be selected so that no security classification is required. Meanings, such as equipment model designation, trade name, military project code name, geographic location, may be used as key words but will be followed by an indication of technical context. The assignment of links, roles, and weights is optional.

Unclassified  
Security Classification